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A-4F Blue Angel Flight Usage Data, 1974-1975

Naval Air Development Center Warminster Pa Air Vehicle Tech Dept

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A-4F BLUE ANGEL FLIGHT USAGE DATA 1974 - 1975

K.I. Leikach and G.A. Bohannon Air Vehicle Technology Department NAVAL AIR DEVELOPMENT CENTER Warminster, Pennsylvania 18974



30 JUNE 1976

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20 ABSTRACT (Continue on reverse side if necessary and	icontify by block number)	This report provides flight
usage data collected and used to cal	culate/monitor	structural fatigue life ex-
pended on individual Blue Angel airc	raft and oresen	ts flight environmental data
which were used to establish full so	ale fatigue tes	t requirements. A total of
4386.0 hours of counting acceleromet	er data (record	ed on a flight by flight ba-
sis) and 436.0 hours of oscillograph	. data (normal &	cceneration, airspeed, and
altitude data) were processed and ar from this data have application in f	e presented. L atique life cal	culations/testing and in the
identification of extreme or excessi	trends detri	mental to prolonging
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SUMMARY

The mission of the Blue Angel Flight Demonstration Squadron is to demonstrate precision techniques of naval aviation. The flight demonstration consists of six aircraft, four "diamond" and two "solo," which perform a prescribed sequence of precise aerobatic and close formation maneuvers at low altitudes. The maneuvers performed represent tactical techniques developed in practice and actual combat. During such maneuvers, pilots and aircraft are repeatedly subjected to high magnitude positive and negative loads or g's. This type of flight loading significantly affects aircraft fatigue life expectancy, and since Blue Angel aircraft were recruited from the fleet where additional fatigue damage was previously accumulated, the need to assess and monitor each aircraft is strongly accentuated.

The objective of the Blue Angel Structural Fatigue Life Monitoring Program, AIRTASK A4101A1/G01-2/6A0400000; sponsored by the Naval Air Systems Comman! (AIR-530), is to maximize operational availability of squadron aircraft without compromising structural reliability. By combining flight usage data with cumulative fatigue damage theory and test results, the structural fatigue life expended for each aircraft is reported throughout its service life. Based on data generated by this program, aircraft may be rotated through the solo and diamond positions to efficiently prolong operational life. Usage data is continuously monitored for adverse trends having a projected impact on aircraft availability.

Reported herein are data representing 4386.0 hours of counting accelerometer reports and 436.0 hours of VGH oscillograph records which were collected during 1974-1975. Load spectra presented have application in fatigue life calculations, in the identification of extreme or excessive exceedance trends detrimental to prolonging fatigue life, and in the development of a loading spectrum for the full scale fatigue test (to establish Blue Angel aircraft fatigue life limits). Additionally, normal acceleration, airspeed and altitude rlight parameters obtained from oscillograph units were used to determine the flight test condition "point in the sky." The "point in the sky" identifies the aircraft gross weight, airspeed, and altitude at which all maneuvers are expected to be performed. These parameters have considerable impact on aircraft wing loading, center of pressure, tail loads, etc., and must be carefully considered in a fatigue test program in order to simulate actual loads as accurately as possible.

This report provides the basic data upon which the MAIRDEVCEN fatigue life monitoring program (i.e., tail-tracking) is mased. No attempt is made herein to evaluate the data for planning projec ed operational service life of squadron aircraft. A follow-on report addressing the structural fatigue integrity of individual airplanes is planned for issue in FY 77.

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INSTRUMENTATION

Blue Angel aircraft were equipped with counting accelerometer and oscillograph recorders during the 1974-75 seasons for the purpose of obtaining normal acceleration, airspeed, and altitude data. Descriptions of these recording devices are contained in Appendix A. Accelerometer groups "counting" normal accelerations at the 5, 6, 7, and 8g levels were installed in all A-4F aircraft (see Figure 1). Oscillograph systems were installed in the diamond "slot" (number four position) and both solo (number five and six positions) aircraft. These three positions were chosen on the basis of the extreme flight loading severity experienced by the two solo positions, and the relative flight load severity of the diamond slot aircraft among diamond positions. Table I contains a list of pilots and the position history of instrumented aircraft during the 1974-75 seasons.

DATA PROCESSING

Counting accelerometer data for each aircraft were submitted monthly by the Blue Angel Squadron using the forms of Figure 2. These reports reflected daily flight by flight information including flight date, flight duration, counting accelerometer readings, and aircraft flight purpose or utilization. Upon receipt at NAVAIRDEVCEN, counting accelerometer data were quality checked for erroneous readings (or component malfunction) and classified into utilization categories. For the purpose of this report, eight utilization categories are identified: solo show, solo practice, diamond show diamond practice, cross country, check-out, other, and unknown. The latter two utilizations were created to account for flights where little or no information was available (i.e., "unknown" refers to unknown flight time in one of the above categories, while "other" refers to flight time which does not fit any of the above).

Oscillograph VGH data, recorded on light sensitive film (called records), were collected by squadron personnel and forwarded weekly to NAVAIRDEVCEN. These records contain a continuous time history of normal acceleration, airspeed, and altitude flight parameters during wheels up flight. Upon receipt, records were logged and flight parameter traces reviewed for clarity and continuity (see Figure 3). Acceptable records were then analyzed to determine scale factors, number of flights on the record, flight duration, and gross weight change during each flight (using fuel consumption and flight maneuver data). Scale factors, (referred to as sensitivities and defined in units of deflection distance per parameter unit, e.g., 6.35 millimetres per "g") were determined initially from oscillograph galvanometer calibrations prior to aircraft installation and were also recorded at the beginning of each record. Serving as a calibration measure of the oscillograph internal circuitry for a particular record, sensitivities were monitored for deviation throughout the "reading" process. Where deviations occurred, adjustments were made to the data based on either previous and subsequent record sensitivities or known information about the parameters (i.e., adjusting the sensitivity to yield approximately the same number of normal acceleration exceedances indicated by known counting accelerometer data).

"Reading" of records was accomplished using electronic, semi-automatic equipment on a flight by flight basis. Only peak values of normal acceleration (ng) above + 2.0g and below + 0.2g were read along with simultaneous values of gross weight, airspeed, and altitude. More than one ng peak per maneuver was read if the trace returned to a position at least one-half the distance of the previous peak (e.g., if a positive maneuver of 5.0g was read followed by a positive 3.0g reading, then somewhere between, the aircraft must have experienced + 2.5g or less). Values of normal acceleration, gross weight, airspeed (knots) and altitude (feet) in order of occurrence were classified by vircraft and placed on computer tapes referred to as listings.

Together with each oscillograph record, the Blue Angel Squadron forwarded pilot flight reports (Figure 4) to aid in the clarification and verification of oscillograph data. Containing counting accelerometer readings and other pertinent flight information, these reports were completed after each flight and corresponded to flights on the oscillograph record. Upon receipt at NAVAIRDEVCEN, pilot flight report data were matched to both oscillograph flight data and flight data from the counting accelerometer reports (Figure 2) as a check to insure accuracy. It is noted that flight duration values as indicated on the pilot flight reports and counting accelerometer reports reflect time between takeoff and landing. Actual oscillograph flight duration values, on the other hand, reflect the time that the oscillograph is operational (i.e., landing gear retracted; wheels-up) and therefore do not include taxi time or the time for those solo and diamond maneuvers which are performed with landing gear extended.

DISCUSSION

Counting Accelerometer

"Counting" positive normal accelerations or maneuver load exceedances between + 5.0g and + 8.0g, accelerometer groups provided 1782.1 and 2603.9 hours of acceptable data in 1974 and 1975, respectively, (Tables II and III). For each aircraft and year, data obtained from the monthly records (Figure 2) were classified into the eight utilizations previously mentioned and are presented in Appendix B. These cumulative exceedance data provided the basis for the + 5.0g to + 8.0g contribution to fatigue life expenditure in accordance with the fatigue analysis of reference (a). Further, these data and corresponding flight hours formed the basis for the development of an 5.0g to 8.0g exceedance rate per 1000 hours for each utilization. Tables IV and V summarize the "g" load history for each squadron aircraft. Cumulative exceedance rates/1000 hours for each utilization category are also shown. The "other" and "unknown" utilization rates are not computed, since they are not normally scheduled missions, but serve as categories for data about which little or no information is available. Graphically, the utilization cumulative exceedance rates are plotted in Figures 5 through 10. Figures 11 thru 15 compare 1974 versus 1975 rates for solo show, solo practice, diamond show, diamond practice, and cross country flights. Table VI contains cumulative exceedance rates per 1000 flight hours of the combined 1974-75 period. It is noted for graphical purposes, all exceedance ranges are plotted at the range mid point (e.g., range 5-5.99 plotted at 5.5). Additionally, where plotted points did not fall on a smooth curve, the curve was faired. Exact points may be obtained from the appropriate table.

OSCILLOGRAPH

To obtain maneuver load spectra in both the positive and negative load regimes, 157.1 and 278.9 hours of acceptable oscillograph data were collected during the 1974 and 1975 seasons, (refer to Table VII). Table VIII presents actual cumulative exceedance data for each oscillograph instrumented aircraft (e.g., aircraft no. 154986 flew 63.02 flight hours and experienced no negative g's below - 3.0, eight occurrences below - 2.00, thirty one occurrences below - 1.00, etc.). To develop cumulative g count data for each specific utilization, the following criteria were applied to the data:

- 1. Only oscillograph flights with matching pilot flight reports (Figure 4) were selected. Additionally, pilot flight reports also must have agreed with corresponding counting occelerometer reports (Figure 2).
- 2. Only solo show, solo practice, diamond show, diamond practice, and cross country flights were selected. The relative importance of chese utilizations and the lack of oscillograph data in the check out utilization were responsible for this criterion.

Table IX presents a list of instrumented aircraft and the amount of available oscillograph data after the above criteria were applied. Data are presented by year, aircraft, and mission utilization (solo show, solo practice, diamond show, diamond practice) in Table X. All data were grouped by mission category and are presented in Table XI. Exceedance rates/1000 hours are given in Table XII. Cross-country flight data, as indicated in Table XIII, reflect data from all instrumented aircraft of Table IX. Comparisons of 1974 versus 1975 oscillograph data for each category are shown in Figures 16 through 20. Figures 21 through 25 graphically present combined 1974-75 oscillograph and counting accelerometer cumulative utilization exceedance rates taken from Tables VI, XII, XIII (check out utilization excluded).

Based on data contained in this report, the following are noted:

- 1. Solo and diamond practice spectra are lower than their respective show rates.
- 2. Solo show exceedance rates at all g levels are the most severe. The maximum positive and minimum negative g's recorded (+ 9.19 and 4.39 in 1975) occurred during solo show flights.
- 3. Solo and diamond practice utilization rates obtained from oscillograph data do not reflect the extensive practice sessions between January and March since oscillograph units were not operational during this period. (See dates of installation in Table VII). Consequently, small differences result between oscillograph and counting accelerometer exceedance rates.
- 4. The time lapse between aircraft acceptance in November 1973 and counting accelerometer installation in March 1974 accounts for the large amounts of "unknown" usage recorded.

- 5. Diamond show and diamond practice counting accelerometer data reflect all diamond position aircraft, as opposed to oscillograph data which reflect diamond slot usage only.
- 6. Solo show and solo practice exceedance rates indicate a significant increase in frequency of negative g occurrence in 1975 from 1974. Diamond show and diamond practice exceedance rates indicate a significant decrease in negative g occurrence in 1975 from 1974.
- 7. At "g" levels common to both the oscillograph and counting accelerometer (+5.0g to + 8.0g):
- a. 1974 solo show and solo practice rates are generally in good agreement with 1975 data.
- b. The slot diamond airplane in 1975 flew less severely in shows than in 1974 (Figure 18 and Table XII). This decrease is not notable in Figure 13 which indicates the "g" count rate of diamond show aircraft flight as an aggregate did not significantly change from 1974 to 1975.
- c. Diamond practice rates (from Figures 14 and 19, Tables IV, V, and XII), reflect a significant increase in 1975 from 1974.
- 8. Cross-country exceedance rates (from Figure 15) generated from counting accelerometer data indicate little or no change from 1974 to 1975. Oscillograph rates (Figure 20) indicate an increase in frequency at the negative g levels in 1975 from 1974. Differences between counting accelerometer and oscillograph rates (especially at the + 5.0g to + 7.0g levels in Figure 25) are due to the fact that oscillograph rates are influenced almost entirely by solo aircraft.
- 9. The utilization exceedance spectra of Figures 21-25 and Tables VI, XII, and XIII represent the complete positive and negative (oscillograph and counting accelerometer) maneuver load exceedance rates for solo show, solo practice, diamond show, diamond practice, and cross-country for combined 1974-75. These rates were based on the greatest amount of available data and together with the counting accelerometer data of Appendix B provide the spectra/data required to project/calculate individual aircraft fatigue life. These spectra also serve as a comparison measure to identify excessive or extreme exceedance trends detrimental to fatigue life.

In an effort to verify counting accelerometer calibration levels and provide a check on total "g" count exceedances as measured by both the accelerometer and oscillograph, the data of Table IX was compared with corresponding counter reports. For each aircraft, oscillograph exceedance data was classified into counting accelerometer calibration ranges obtained from Table A-I, Appendix A (e.g., counter calibration ranges for aircraft 154177 were 5.09-6.07, 6.08-7.11, 7.12-8.10, 8.11 and up). Dates of transducer (range) changes were applied to the oscillograph data appropriately. Using only solo show, solo practice, diamond show, and diamond practice utilizations (from the data of Table X), the results of this comparison are indicated in Table XIV. Exact verification of counting accelerometer and oscillograph exceedances was not obtained; however, the oscillograph values were reasonably approximate at all g levels (or calibration ranges) to be considered a good indicator of exceedance trend. Discrepancies may be due to a combination of inconsistencies in oscillograph calibration techniques, and small inherent human and machine error common to oscillograph "reading" processes.

In addition to recording maneuver load (normal acceleration) histories, oscillograph units provided a continuous time history of aircraft airspeed and altitude. These parameters can affect the critical flight condition determination for full scale fatigue tests and analyses by influencing such factors as wing center of pressure, wing load distributions, tail loads, etc. The fatigue test critical flight condition "point in the sky" determines the airspeed and altitude at which all maneuvers are expected to occur. To provide a survey of airspeed, altitude, and normal acceleration parameters to be used in full scale fatigue test loads considerations, the osc: llograph data indicated by Table VII (pilot report sheet match not required) were classified by solo or diamond aircraft position for the combined 1974-75 seasons. Altitude and airspeed versus "g" exceedance data are presented in Appendix C, indicating "where" recorded maneuvers occur. Altitude bands chosen range from below sea level, -609.6 metres (-2000 ft.). to 9143.7 metres (30,000 ft.) for airspeed values ranging from 100 to 699 knots in increments of 50 knots. Also indicated is the average aircraft gross weight during maneuver exceedance for the altitude band indicated. Table XV summarizes this data in terms of percent of exceedances occurring below various airspeed values in altitude bands for diamond and solo aircraft. This table indicates:

- a. For diamond utilization, 92.6% of all exceedances occurred below 1523.7 metres (5,000 feet) with 99.3% of these occurring at less than 500 knots.
- b. For solo utilization, 87.2% of all exceedances occurred below 1523.7 metres (5,000 feet) with 99% of these occurring below 500 knots.



Figure 1. Blue Angels

"BILLE ANGFLS" COUNTING ACCELEROMETER REPORT

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Figure 2. Blue Angels Counting Accelerometer Report

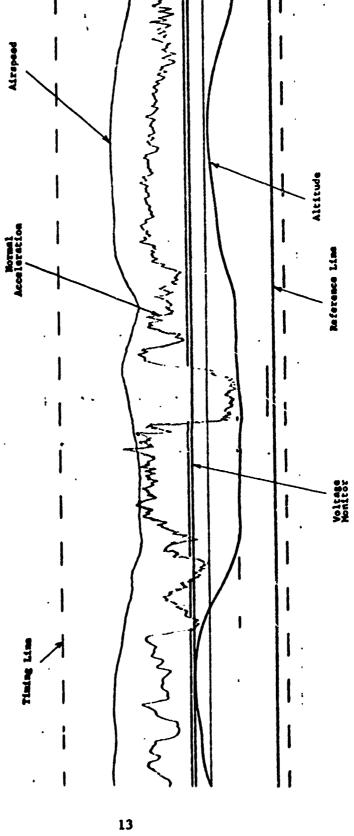


Figure 3. Oscillograph Trace

DATE

PILOT FLIGHT REPORT

NASC AIR TASK 10400301/NJ201 "Blue Angels" A-/F Aircraft Structure Integrity

TO THE PILOT: This aircraft is equipped with flight recorders which automatically produce a record of airspeed, altitude, and flight accelerations. This information will contribute to improvement of structural design requirements and fatigue life determination. The following information is required for proper evaluation of the recorded data. Your assistance is greatly appreciated.

"BLUE ANGELS"

NAVAL AIR DEVELOPMENT CENTER

WARMINSTER, PA 18974

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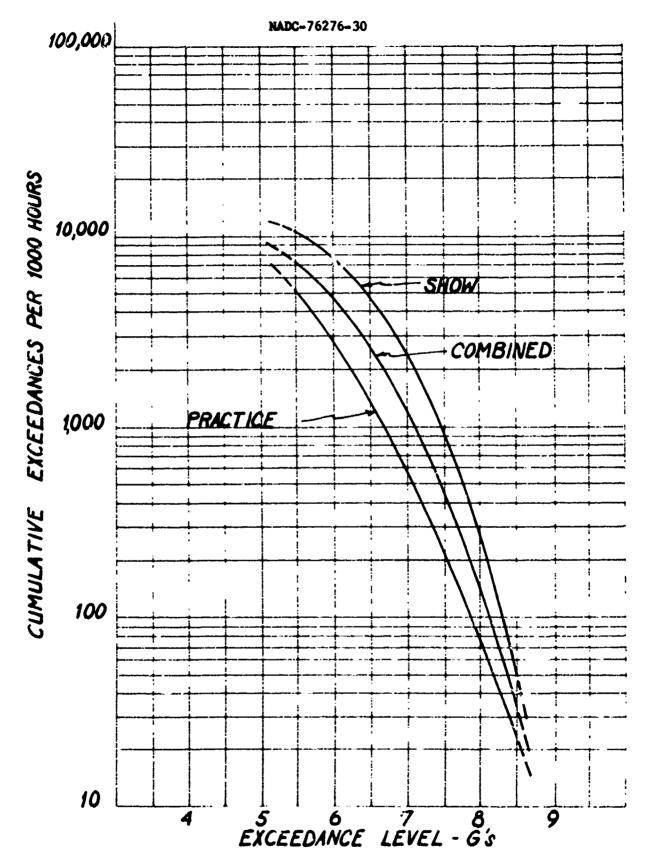


Figure 5. Blue Angel 1974 Counting Accelerometer dates per 1000 Hours - Solo Aircraft

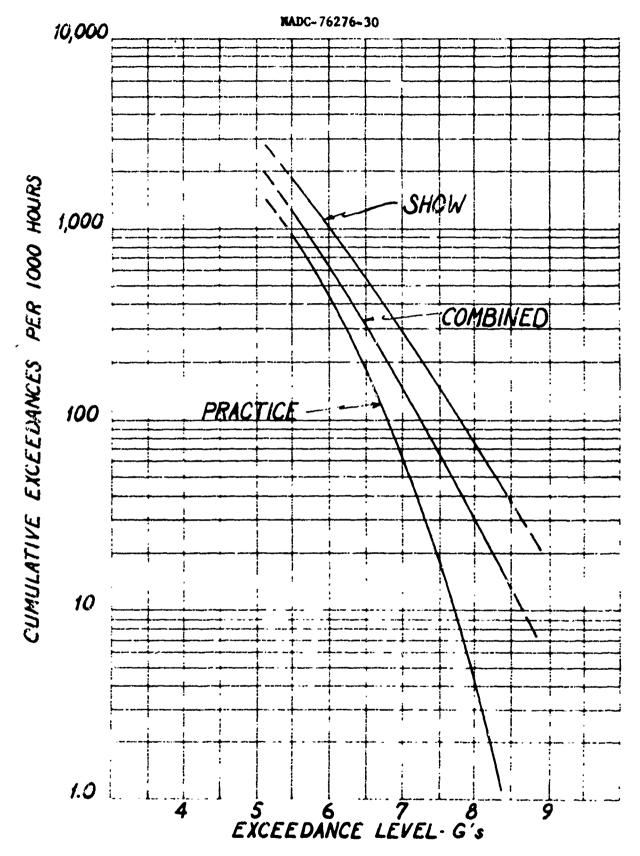


Figure 6. Blue Angel 1974 Counting Accelerometer Rates Per 1000 Hours - Diamond Aircraft

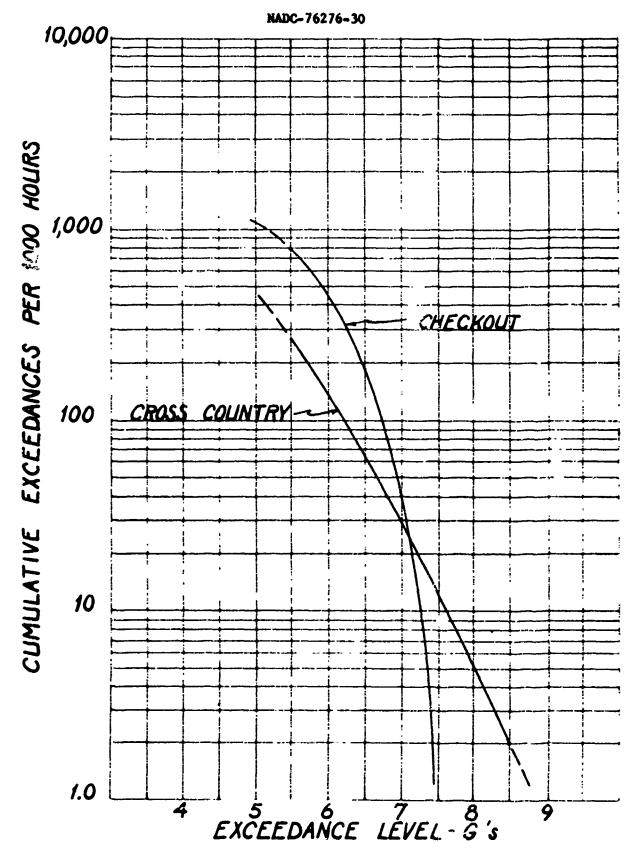


Figure 7. Blue Angel 1974 Counting Accelerometer Rates per 1000 Hours - Cross Country - Checkout

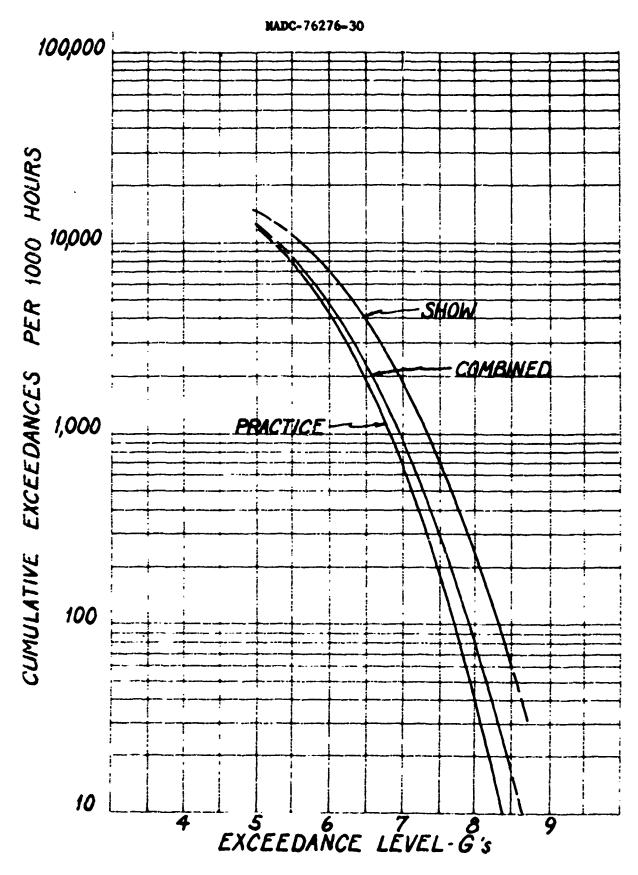


Figure 8. Blue Angel 1975 Counting Accelerometer Rates Per 1000 Hours - Solo Aircraft

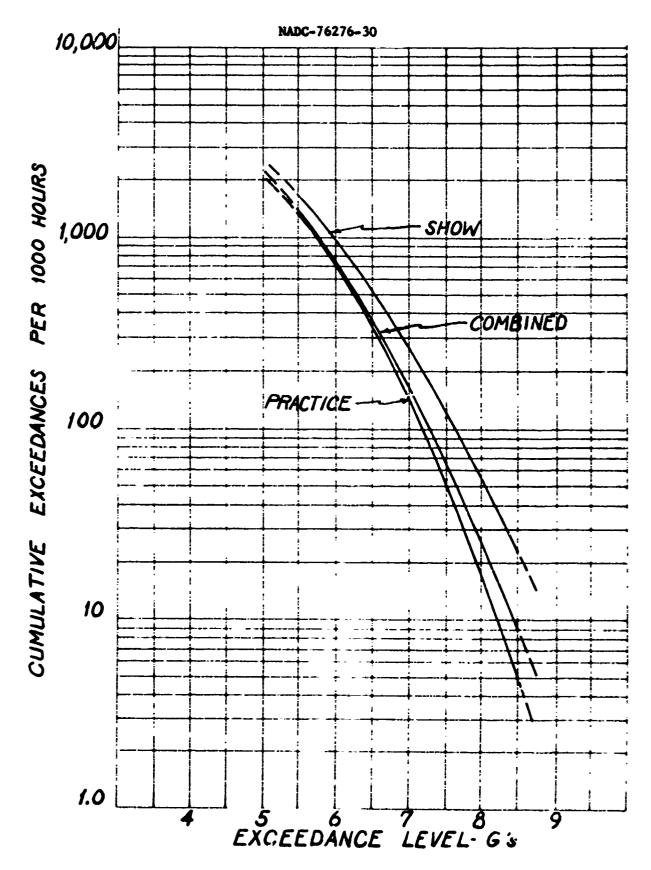


Figure 9. Blue Angel 1975 Counting Accelerometer Rates per 1000 Hours - Diamond Aircraft

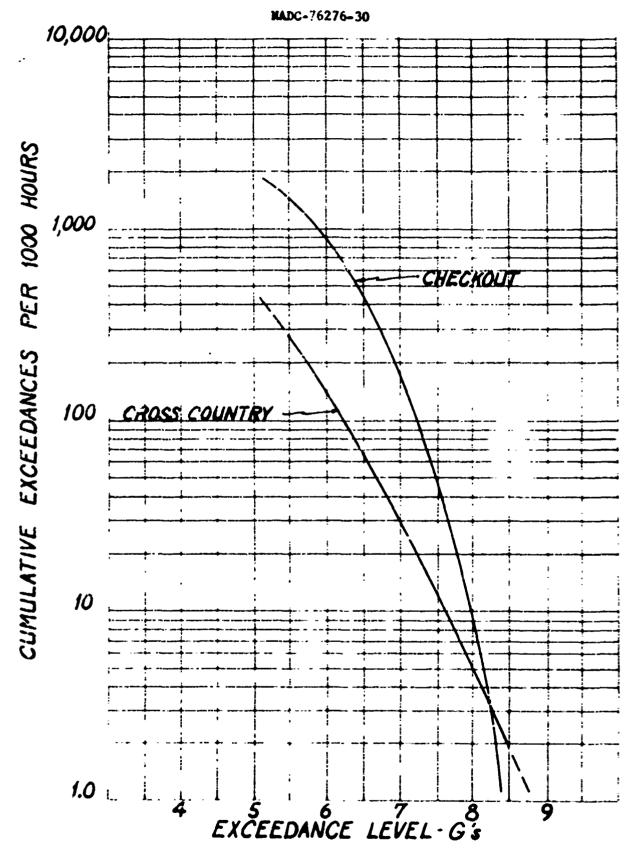


Figure 10. Blue Angel 1975 Counting Accelerometer Rates Per 1000 Hours - Cross Country - Checkout

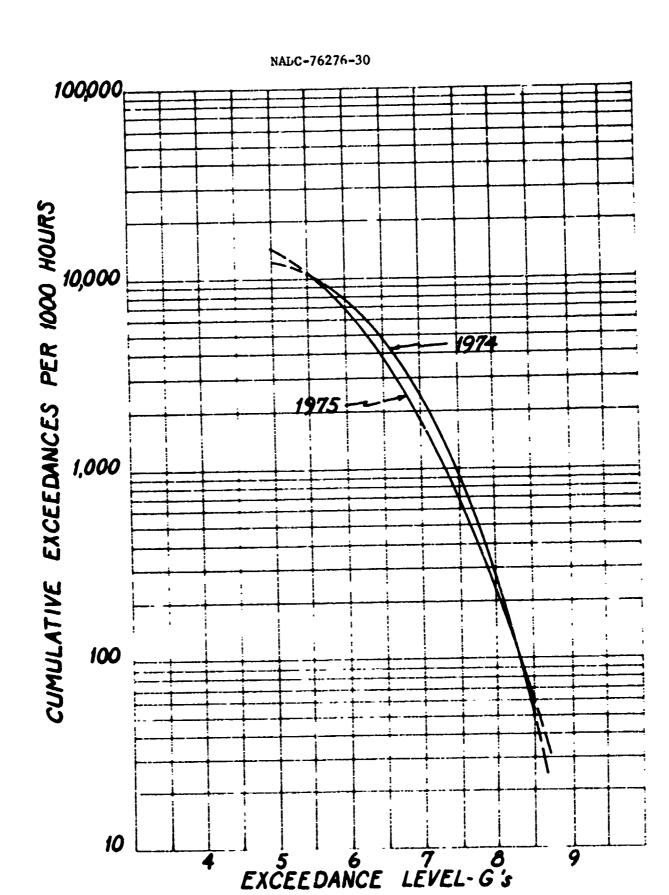


FIGURE 11. 1974 vs. 1975 Blue Angel Counting Accelerometer Rates Per 1000 Hours - Solo Show

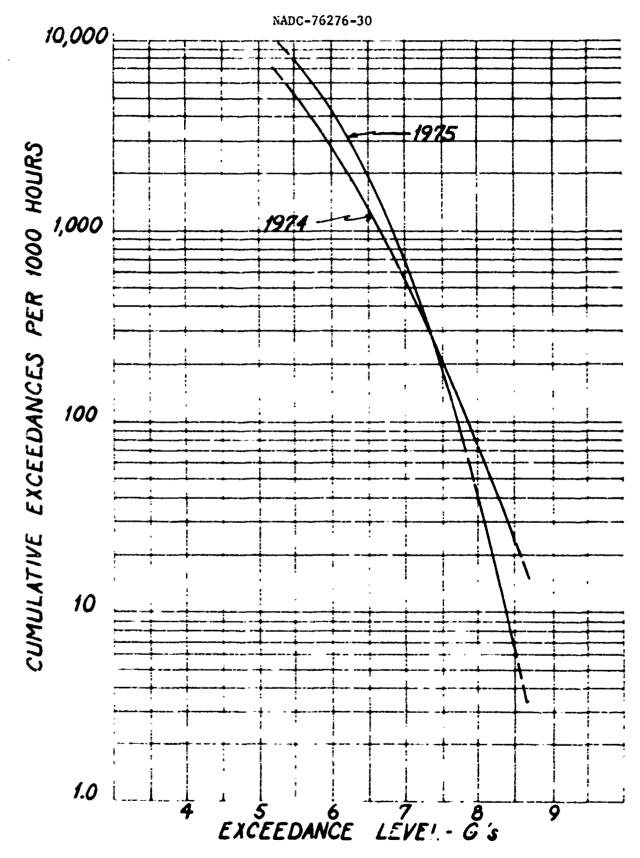


FIGURE 12. 1974 vs. 1975 Blue Angel Counting Accelerometer Rates Per 1000 Hours - Solo Practice

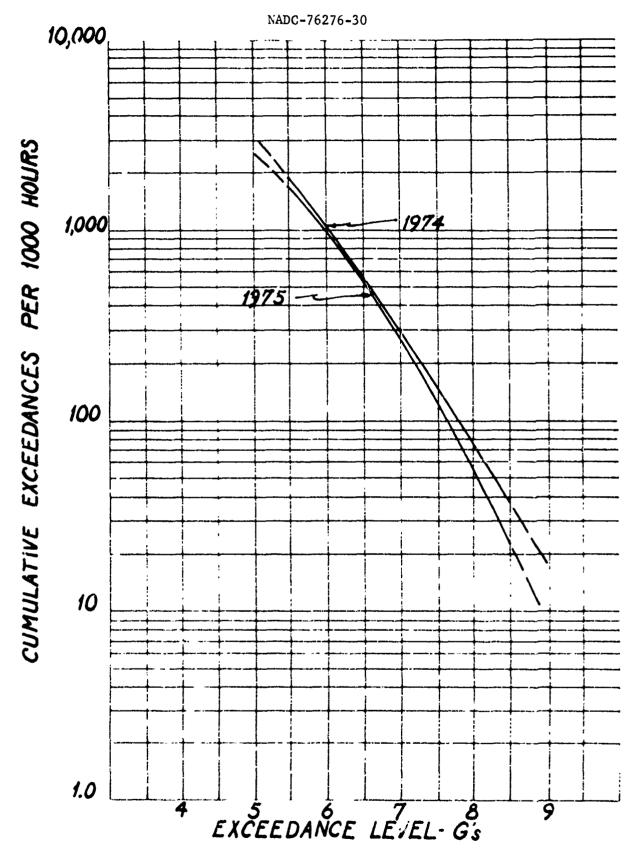
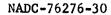


FIGURE 13. 1974 vs. 1975 Blue Angel Counting Accelerometer Rates Per 1000 Hours - Diamond Show



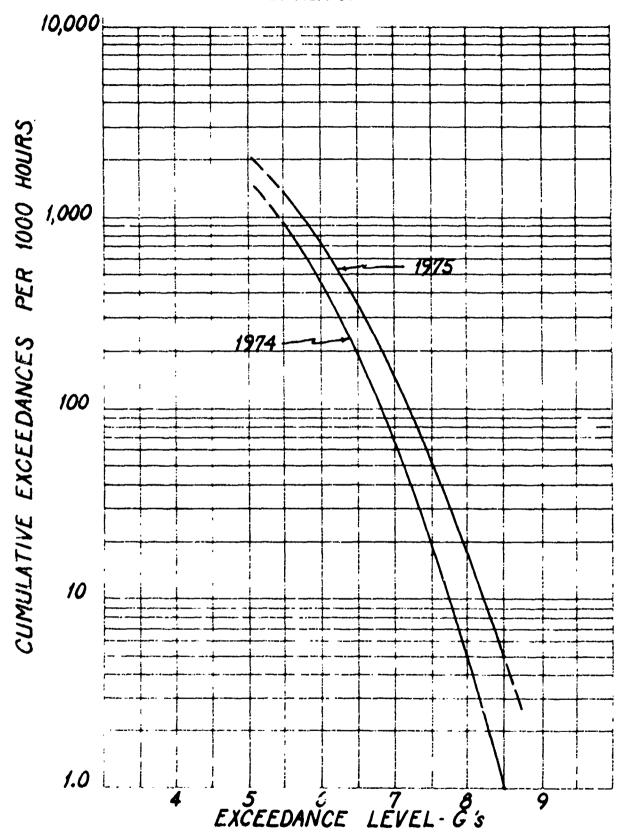


FIGURE 14. 1974 vs. 1975 Blue Angel Counting Accelerometer Rates Per 1000 Hours - Diamond Practice

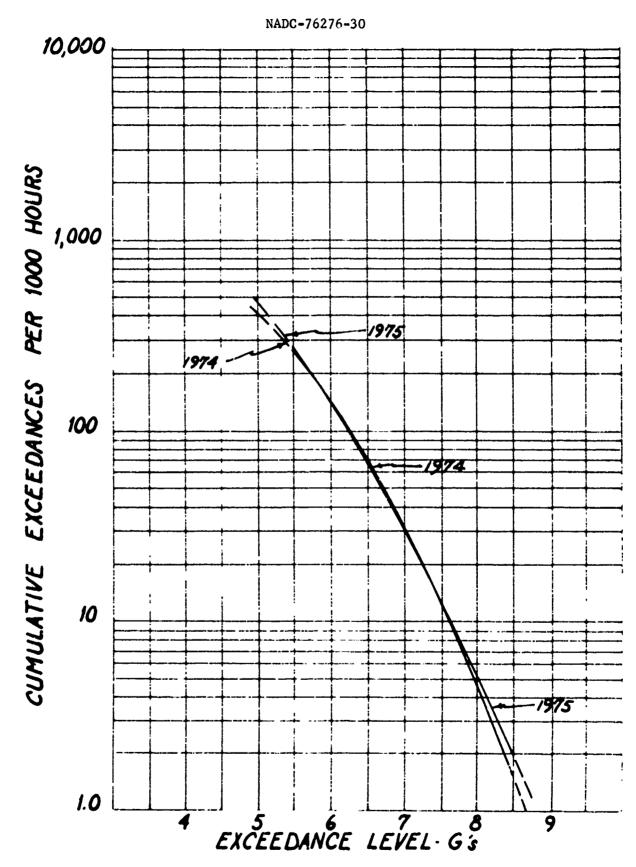
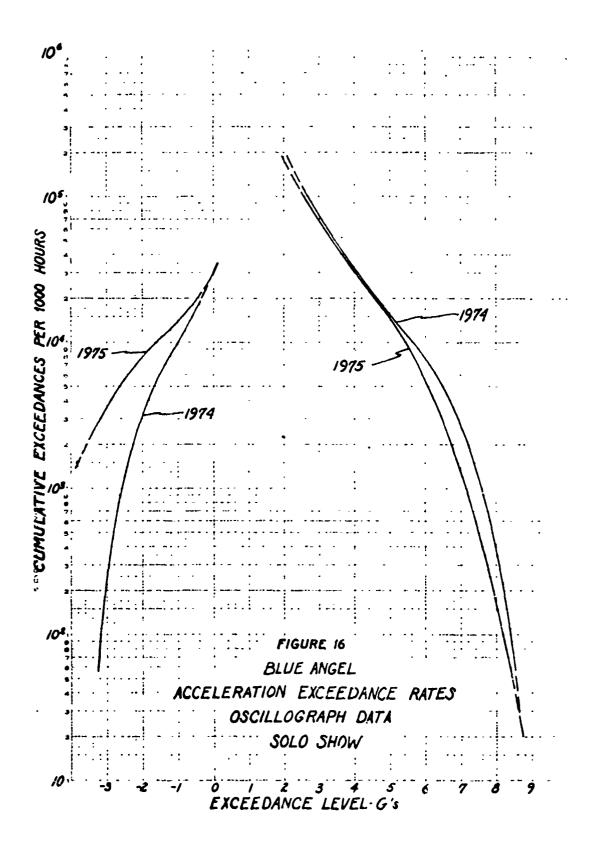
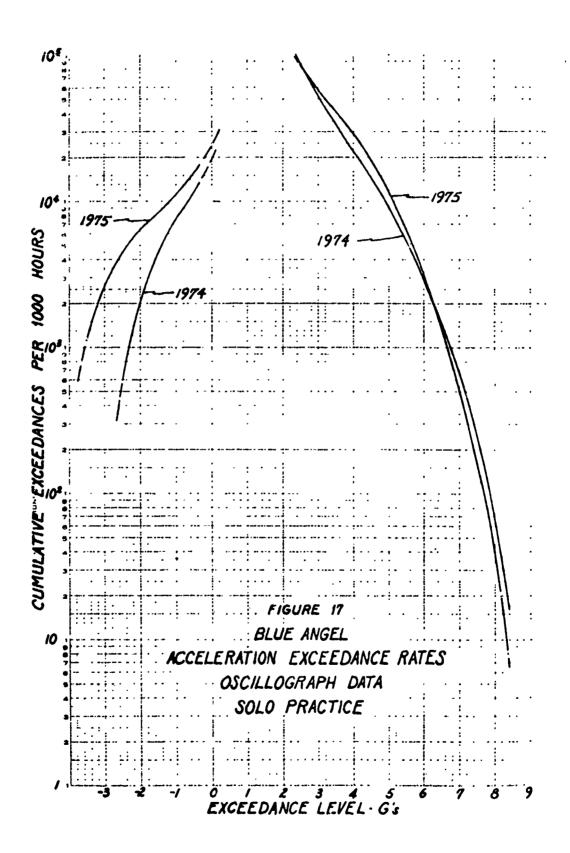
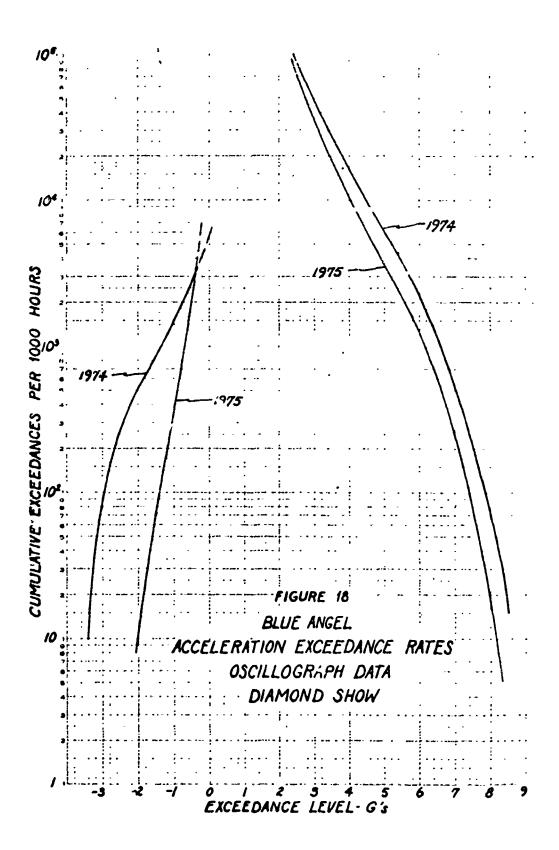
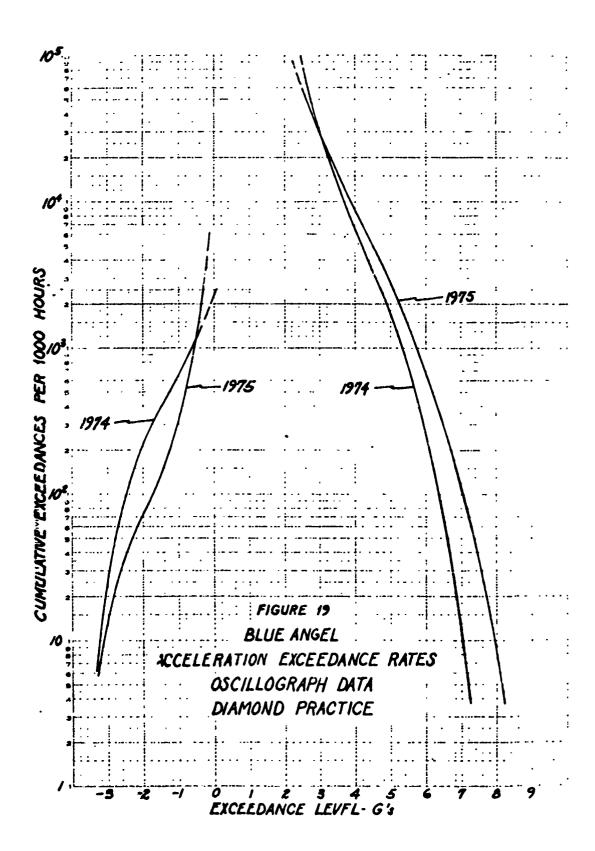


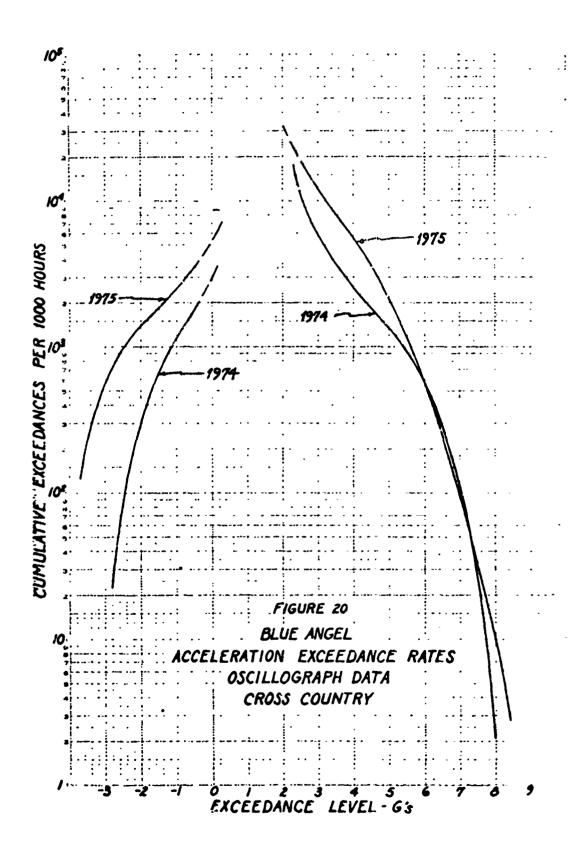
FIGURE 15. 1974 vs. 1975 Blue Angel Counting Accelerometer Rates Per 1000 Hours - Cross Country

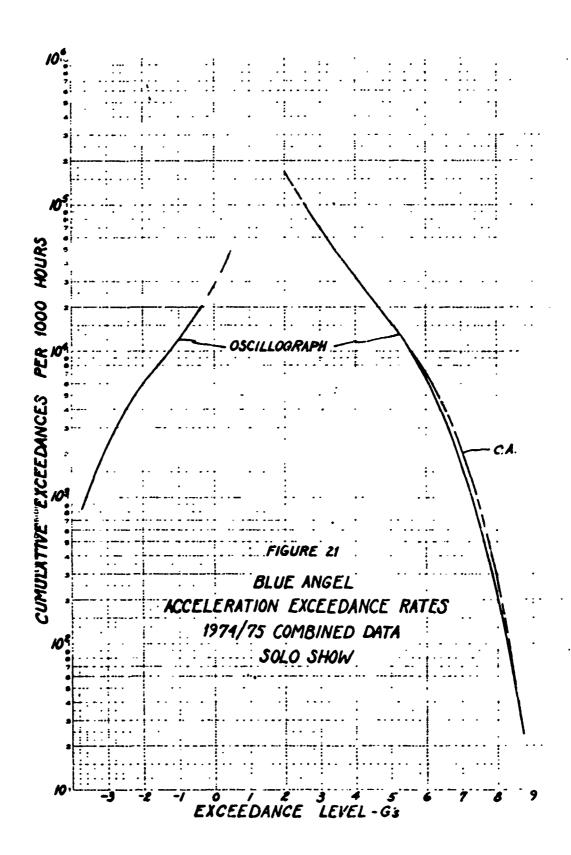


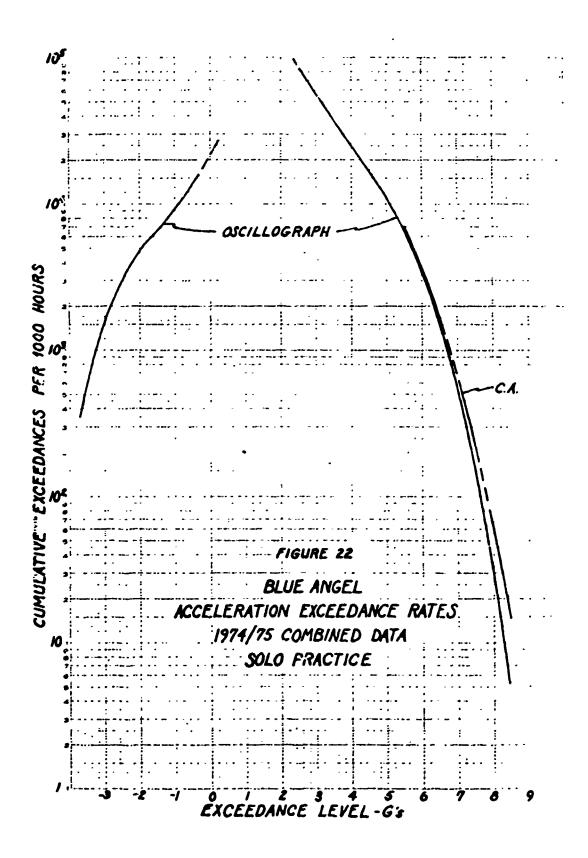


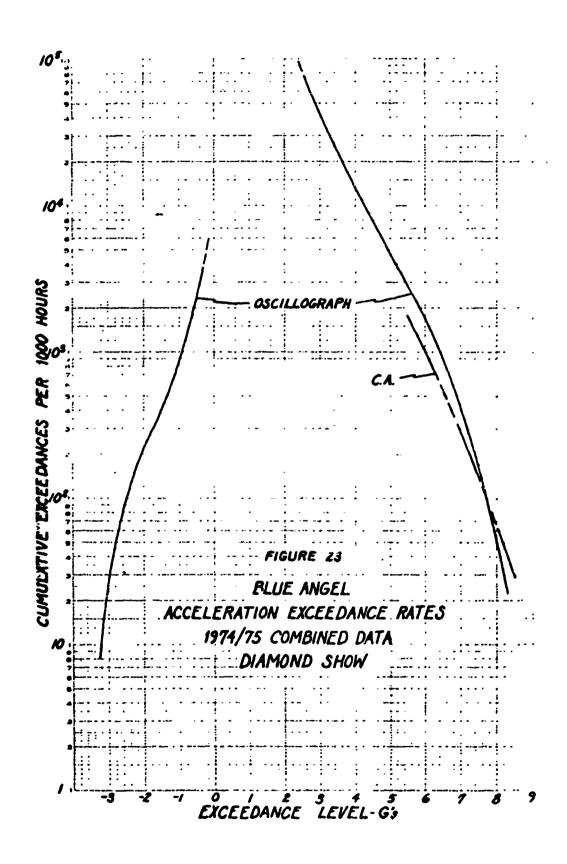


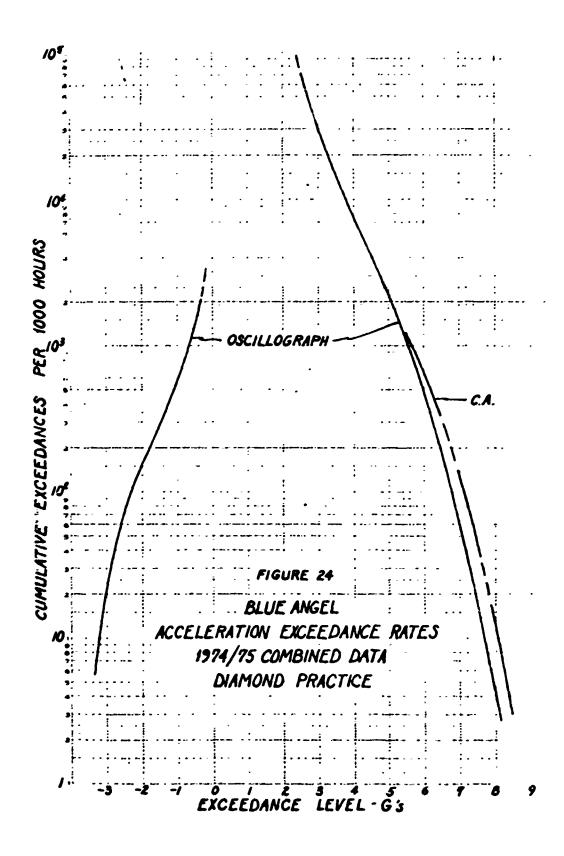












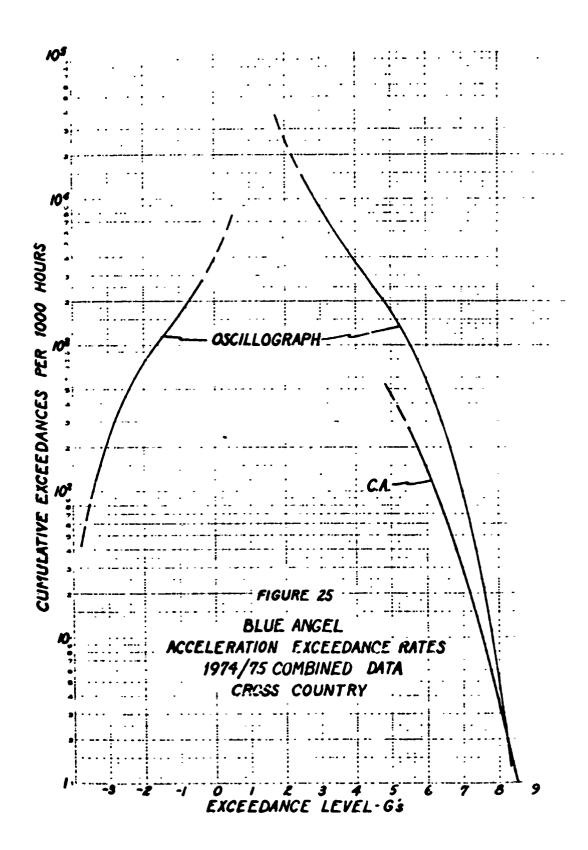


TABLE 1

1974-1975 BLUE ANGEL PILOTS AND AIRCHAPT POSITION HISTORY

			150 150 150 150 150 150 150 150 150 150	ATRCRAFT POS IT ION					ĕ	1974 CDR. A. LESS	SST			•	1975 CDR. A. LESS	있. 3	es es					
			14	~					1001	ن ند	LCDR. M. WIITA				S.	3 E	CAPT. W. HOLVERSTOTT	TOT				
-			•	_			•		:	د. 9	LT. J. CHEILMISKY	⋧			LT. J. PATTON	. PAT	TON					
			.1						2	CAPT. J. FOGG	FOCC				LT. J	. Cits	LT. J. CHEHAMSKY	-				
			•	_					11.	LT. J. TUCKER	JCKER			_	LT. V. PARKER	. PAR	KER					
			•	<u>,</u>					7	LT. V. PANCER	NAKER			_	LT. D. SAPP	. SAP	۵.					
			^	(ne.	(narrator)	^			.	LT. J. PATTON	VTTOX.			_	LT. A. CISNEROS	. C1S	NEROS					
AIRCRAFT SERIAL							i					 		I	-							
NO.	,	•	x	<	x 15%	72	~	<	••	0	z		7	~	r	<	H 1975	داد د	<	•	0	
154176	•	•	=	•	•	•	4/2	2/8	•	-	•	•	Į	4	•	•	•	9	•	•	•	
154177	•	•	\$	•	1/6	1/5	8/8	9/6	•	•	•	•	•	9/ 0	-	•	8/5	~	~	~	~	
154179	•	•	•	•	-	•	•	\$	•	•	•	•	į	۵	•	6	 n	~	•	•	~	
154975	•	•	~	~	m	•	•	3/9	~	۳,	•	•	A	۵	4	•	•	4	4	•	4	•
154983	•	٠	~	~	~	~	2/8	8/2	~	~	~	•	<u>-</u>	8/ 0	•	•	1/6	_				
154984	•	•	-		\$	\$	~		~	_	-	•	 _	S/ Q		_	6/1	•	5/9	•	•	-
154980	•	•	•	•	4	4	*	4	•	4	•	•	a	Ω.	~	~	2/8 x	••	••	•	•	• • •
155029	•	•	•	~	~	\$	\$/8	~	~	•	~	•	s	-/8	~	~	8	•	9/8	•	•	•
* 68. 69 - SPARE	•	Ade																				

ABLE II

BLUE ANGEL COUNTING ACCELEROMETER DATA SUMMARY

FLIGHT

တ	,1	7	8	0	0	0	7	0	
EXCEEDANCES	23 1	372 103	131 16	15 1	6 65	2	81 27	237 19	
ы Ж		ю	7					7	
νl	142	723	767	160	92	93	288	932	
ACCEPTABLE COLNTING ACCELEROLETER	156.0	228.0	160.2	244.3	245.6	245.8	258.7	243.5	1782.1
BLUE ANGEL	203.3	340.1	300.3	280.5	371.3	365.5	271.6	286.2	7418 8
TOTAL	1912.8	2409.1	2279.9	2941.8	2440.7	2164.8	1744.9	1982.5	
SERIAL NIN 3ER	154176	154177	154179	154975	154983	154984	154386	155029	

TABLE III

1975

SILE LYGEL CHUNTING ACCELEROGETER DATA SUNDARY

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⊢ 1	I
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જા	0	4	-	-	0	0	9	7	
EXCEEDANCES	599 43	69 16	131 25	151 22	32 1	104 17	132 22	. 426 58	
νl	2065	352	507	509	149	447	, 421	1492	_
ACCEPTASIE COUNTING ACCELEROMETER	325.7	302.5	413.0	410.0	285.2	325.2	257.0	285.3	2603 9
BLUE ANGEL	394.3	327.8	413.0	410.0	285.2	325.2	257.0	285.3	2697 8
TOTAL	2307.1	2736.9	2692.9	3351.8	2725.9	2490.0	2001.9	2267.8	
SERIAL NUMBER	154176	154177	154179	154975	154983	154984	15-986	155029	

TABLE IV

1974 COUNTING ACCELEROMETER UTILIZATION SUMMARY

	8010	O S H	* 0			0708	P R A	CTI	S S		CROSS	0 0 C	H Z	× ×	
	ACCEPTABLE HOURS	~	CUMULATIVE EXCEEDANCE:	CUMULATIVE EXCEEDANCES 6 7	60	ACCEPTABLE HOURS	8	CUMU EXCE 6	CUMULATIVE EXCEEDANCES 6 7	ю.	ACCEPTABLE HOURS	^	CUMO EXCE	CUMULATIVE EXCEEDANCES 6 7	, •
154176	٥٠٦	15	8	0	0	14.1	37	es	0	0	83.1	23	S		-
154177	33.8	426	242	20	e	24.9	191	76	27	4	122.6	89	54	4	0
154179	25.6	242	97	11	8	41.4	194	52	ო	0	65.8	32	9	7	0
154975	•	• •1	•	•	•	•	,	•	•	1	108.9	^	0	0	0
154983	•		•	•	•	3.4	٥	0	0	0	99.1	m	8	0	0
154984	•		٠	•	•	•	•	•	•	,	113.3	9	0	0	0
154986	•	•	•	•	•	•	'		•	1	118.8	9	~	~	0
155029	41.7	383	136	2	0	83.6	456	25	€0	0	106.2	22	81	0	0
TOTALS	101.8	1066	084	16	S	167.4	878	197	38	4	817.8	218	57	80	-
RATES PER 1000 HOURS		10472	91.17	894	S		5245 1177	1111	228	24		267	07	10	7

TABLE IN

1974 COUNTING ACCELEROMETER UTILIZATION SUMMARY

	BIAMOR	E S E	, a o x			DIAMONI	0 P R	A C T	I C E		E C	HECKOU	u T		
SERIAL	ACCEPTABLE HOURS	۰	CUMUL EXCEE 6	CUMULATIVE EXCEEDANCES 6 7	60	ACCEPTABLE HOURS	~	CUMU EXCEI	CUMULATIVE EXCEEDANCES 6 7	6 0	ACCEPTABLE HOURS	8	CUMUL EXCEE	CUMULATIVE EXCEEDANCES 6 7	80
154176	11.6	54	'n	0	0	15.8	54	e	0	•	11.2	10	7	0	0
154177	9.7	^	0	0	٥	21.2	16	8	0	0	6.1	6 0	m	0	0
154179	1.7	4	0	0	0	2.0	-	0	0	0	7.7	01	-	0	•
154975	47.2	76	œ	-	0	77.5	8	7	0	0	9.4	0	•	0	0
154983	38.5	39	27	9	0	84.2	20	30	en	•	7.8	0	0	ပ	0
154984	40.9	22	0	0	•	81.9	9	S	0	0	6.5	'	0	0	0
154986	42.6	166	55	23	7	83.0	105	23	m	0	2.5	0	0	0	•
155029	•	,	•	•	•	1.2	4	-	•	0	6.3	•	c	0	0
TOTALS	192.2	356	9.5	30	,	366.8	341	r.	9	0	7.67	39	•	0	°
RATES PER 1000 HOURS		1853	\$67	157	37		930	194	17	0		790	183	0	•

TABLE V

1975 COUNTING ACCELERONETER UTILIZATION SUPPARY

	0108	H S O	3 0			0 1 0 8	4 ×	PRACTICE	en en		CROSS		COUNTRY	> -	
SERIAL	ACCEPTASLE HOURS	~	CUNULATIVE EXCEEDANCES 6 7	CUMULATIVE . EXCEEDANCES 6 7	c o	ACCEPTABLE HOURS	~	CUMUI EXCEE	CUMULATIVE EXCEEDANCES 6 7	6 0	ACCEPTABLE HOURS	~	CUNUL EXCEE 6	CUNULATIVE EXCEEDANCES 6 7	∞
154176	0.97	541	202	23	0	142.1	1394	369	28	0	125.9	122	27	~	0
154177	,	•	•	•	•	23.2	163	23	0	0	122.7	7	-	0	0
154179	•	•	•	•	•	5.2	18	91	4	-	149.1	ដ	е	~	•
154975	,	, 	•	•	•	6.7	09	54	٥٠	0	151.3	10	-	0	0
154983	,	· 	•	•	•	6.0	43	12	7	0	125.8	-	0	0	0
154984	17.8	134	97	10	0	44.7	161	35	4	.0	102.4	54	S	٥	0
154986	1.5	16	•	0	0	20.6	132	43	ю	0	59.5	•	က	7	7
155029	36.5	387	180	36	9	136.0	1023	218	19	-	102.6	75	22	7	0
TOTALS	101.8	1078	767	28	vo		3024	734	\$9	7	939.3	257	62	^	~
RATES PER 1000 HOURS		10590	7925	280	65		7865 1909	1909	170	9		274	67	&	n

TABLZ V

1975 COUNTING ACCELEROMETER UTILIZATION SUMMARY

	DIAHON	D S I	3 O H			DIAMON	۵ ۵	ACT	1 0 1		8 H O	E C K O	T D		
SERIAL	ACCEPTABLE HOURS	\$	CUMUI EXCER 6	CUMULATIVE EXCEEDANCES 6 7	80	ACCEPTABLE HOURS	۰	CUMU EXCE	CUMULATIVE EXCEEDANCES 6 7	. 88	ACCEPTABLE HOURS	۶	CUMULATIVE EXCEEDANCES 6 7	ATIVE MANCES 7	80
154176	•	•	•	•	•	1,3	9	0	0	0	5.0	~		0	0
154177	43.1	. 78	30	12	c	97.6	92	15	4		5.1	4	0	0	0
154179	51.6	102	34	6 0	•	196.0	366	%	12	0	3.7	∞	0	0	0
154975	55.2	70%	33	4	-	192.3	329	%	12	0	1.9	۶.	ю	0	0
154983	43.6	40	10	0	0	.91.9	62	01	,	0	5.4	-	0	0	0
154984	11.7	01	-	0	•	144.8	88	17	е	0	2.2	0	0	•	0
154986	12.7	30	œ	-		150.7	221	79	16	е	3.3	11	•	0	0
155029	•	•	•	•	,	•	•	•	•	1	5.1	^	•	~	•
TOTALS	217.9	370	116	25	~	874.6	1164	280	4.7	4	31.7	45	82	-	0
RATES PER 1000 HOURS		1699	533	115	23		1331 521	321	25	5		1420	568	32	0

TABLE VI COMBINED 1974-1975 COUNTING ACCELERONETER HTILIZATION SUPPRARY

	0108	× 0	n 0 1			0108	7 X Y	PRACTICE	ia C		CROSS		COUNTRY	> -	
SERIAL	ACCEPTABLE HOURS	~	CUNUL EXCEEI 6	CUNULATIVE EXCEEDANCES 6 7	80	ACCEPTABLE HOURS	\$	CIMC EXCEI	CUMUIATIVE EXCEEDANCES 6 7	80	ACCEPTABLE HOURS	~	CIMUL EXCES 6	CURCLATIVE EXCEEDANCES 6 7	€0
154176	46.7	256	207	13	•	156.2	1431	372	78	•	209.0	145	32	n	
154177	33.8	426	242	20	6	48.1	354	1117	27	4	245.3	75	22	4	0
154179	25.6	242	97	11	8	9.97	212	35	7		214.9	45	٥	٣	o
154975	•	•	•	•	,	6.7	09	54	•	0	260.2	15	H	0	0
154983	•	,	•	•	•	7.6	643	12	-	•	224.9	4	8	၁	၁
154984	17.6	134	97	2	0	44.7	161	35	4	0	215.7	30	v	0	0
154986	1.5	91	9	•	٥	70.6	132	43	6	0	178.3	=======================================	•	ы	~
155029	78.2	07.	316	97	•	219.6	1479	293	27	-	208.8	150	04	~	0
TOTALS	203.6	2144	916	150	n	551.9	3902	931	103	9	1.727.1	475	911	15	n
RATES PER 1000 HOURS		10530	10530 4489	736	*		7070 1687	1687	187	u		270	67	80	

TAMIR VI COMBINED 1974-1975 COUNTING ACCELERONETER UTILIZATION SUNDARY

DIAM	0 % 0	×	3			DIANON	D P R	RACTI	3 C I		N O M E C K	~ ~ ~	0 U T		
	ACCEPTABLE POUGS	•	CUNULATIVE EXCEEDANCE 6 7	CUNULATIVE EXCEEDANCES 6 7	40	ACCEPTABLE HOURS	\$	CUMP EXCEE	CUMULATIVE EXCEEDANCES 6 7	•0	ACCEPTABLE) HOURS	8	CUMPLATIVE EXCEET ANGES 6 7	TIVE	==
11.6		54	vs.	0	0	17.1	30	e	0	0	16.2	12	٣	•	0
\$2.8		91	30	12	е.	118.8	108	11	4	,-4	11.2	12	٣	0	•
53.3		106	34	®	•	198.0	367	78	12	•	8.1	18	~	0	0
102.4		198	41	v	~	269.8	410	26	13	•	6.5	ø	•	0	0
82.1	<u>-</u>	42	37	9	٥	176.1	112	7	6	ю	13.2		0	. ح	3
52.6		32	-	0	0	226.7	148	22	е	0	6.7	v	•	0	ပ
55.3		196	63	5%	••	233.7	326	83	19	е	8.8	17	&	0	•
		•	•	•	•	1.2	3	~	0	•	11.4	2	٠	~	0
410.1		726	211	\$\$. 12	1241.4	1505	351	23	4	81.1	78	52	-	0
1	1,1	0771	\$15	134	29		1212	283	43	m	·	1035	33	21	•

TABLE VII
OSCILLOGRAPH DATA SUMMARY

<u>YEAR</u>	SERIAL NUMBER	AIRCRAFT POSITION	ACCEPTABLE OSCILLOGRAPH DATA HOURS	DATA <u>PERIOD</u>	TOTAL NO.
1974	154986	#4(DIAM)	63.02	3/29/74-11/10/74	1
1974	155029	#5(SOLO)	35.66	4/5/74-8/25/74	157.1
1974	154179	#6(SOLO)	38.21	3/29/74-7/7/74	
1974	154177	#6(SOLO)	20.21	9/20/74-11/10/74	
1975	154975	#4(DIAM)	97.87	3/1/75-10/29/75]
1975	155029	#5(SOLO)	44.6	4/11/75-10/27/75	278.9
1975	154176	#6(SOLO)	136.43	2/26/75~10/29/75	
			436.0		

TABLE VIII
OSCILLOGRAPH DATA
CUMULATIVE EXCEEDANCE SUMMARY

	*****	1	974 			1975	
G LEVEL			(S)	154179 (S)	154975 (D)	155029 (S)	154176 (S)
LESS THAN -3.00	0	0	0	0	0	30	165
-2.99 TO -2.00	8	21	23	36	1	94	681
-1.99 TO -1.90	31	149	124	192	5	278	1185
99 TO .25	122	479	380	608	286	707	2347
2.00 TO 2.99	558 ย	3608	1527	4005	4503	2880	10921
3.00 TO 3.99	1530	1569	961	1672	1317	1480	5882
4.00 TO 4.99	384	581	442	673	396	669	2602
5.00 TO 5.99	112	255	225	265	117	269	1082
6.00 TO 6.99	23	84	131	67	31	91	315
7.00 TO 7.99	6	7	42	7	3	16	21
9.00 AND UP	0	0	1	1	0	3	0
HOURS OF DATA	63.02	35.66	20.21	38.21	97.87	44.60	136.43

TABLE IX
OSCILLOGRAPH DATA SUFFMARY: PILOT REPORT SHEET MATCH

YEAR	SERIAL NUMBER	AIRCRAFT POSITION	PRACTICE	SHOW	TOTAL
1974	154986	#4(DIAM)	30.8	19.6	50.4
1974	155029	#5(S0TD)	10.1	14.3	24.4
1974	154179	#6(SOLO)	17.6	9.9	27.5
1974	154177	#6(SOLO)	4.1	11.0	15.1
1975	154975	#4(DIAM)	25.1	27.8	52.9
1975	155029	#5(SOLO)	13.8	12.5	26.3
1975	154176	#6(SOLO)	<u>29. 1</u>	33.1	62.2
			130.6	128.2	258.8

TABLE X OSCILLOGRAPH DATA AIRCRAFT UTILIZATION CUMULATIVE EXCEEDANCE SUHHARY

					Ξ:	1974						1975	۲: د د	•	
6 LEVEL	ڙ	154	154986	155	920	154	177	154179	179	154	154975	155029	620	154	2
	•	PAAC	SHOW SHOW	SOLO SOL PRAC SHO	800 8404	SOLO SOL PRAC SH	SHOW	SOLO SOL PRAC SHO	50L0 SH0%	PRAC	SHOW	SOLO	SHOW	SOLO SOL PRAC SHO	\$01.0 \$10.0
LESS THAN -3.00	-3.00	•	•	•	•	•	•	•	•	•	•	•	5	\$	92
-2.44 70 -2.00	2.00	•	. v	•	=	•	13	~	•	-	•	32	ħ	?	122
-1.99 10 -1.00	-1.00	~	2	92	2	32	2	•	19	•	~	95	3	569	390
00 10	.25	35	3	2	526	•	228	244	180	~ :	111	212	152	537	727
2.00 TO	5.99	506	1658	•7•	1637	30	616	1721	1211	1430	1953	988	996	5494	3630
3.00 10	3.99	497	159	322	179	228	612	999	\$36	415	919	474	495	1309	1907
4.00 10	•••	105	197	109	290	95	205	258	508	129	101	210	223	265	738
5.00 TO	8.44	2	2	;	130	80	142	16	106	37	35	•	123	219	310
0.00 10	•••	~	20	•	;	ĩ	:	•	~	•	:	•	•	;	•
7.00 TO	7.99	•	•	•	•	~	*	-	•	-	~	•	•	•	•
8.00 AND	5	•	•	•	•	•	-	•		•	•	•	~	•	•
MOURS OF DATA	DATA	30.0	19.6	10.1 14.3	14.3	:	0.11	17.6	6.6	25.1	87.8	13.0	12.5	29.1	33.1

TABLE XI OSCILLOGRAPH DATA UTILIZATION CUMULATIVE EXCEEDANCE SUMMARY

			NAD	C 76	276-	30							
	\$00 \$00 \$40#	100	297	769	1639	4563	4359	1745	=	322	88	•	•
1974-1975	PRACTICE	\$	212	205	1167	4229	1000	1272	453	. 101	18	•	74.7
1974-	MONO SHOW	•	S		177	3611	1267	378	130	39	•	•	4.7.4
	PRACTICE SHOW PRACTI	•	•	15	*:	3936	606	\$34	19	•	-	•	\$5.9
	AONS SAON	••	255	463	910	. 9654	2002	196	+33	•••	23	~	*5.6
1975	PRACT ICE	\$	193	361	749	3449	1763	0.10	283	Š	~	•	42.9
	i	•	•		==	1953	919 .	101	\$	2	~	•	27.8
	PRACTICE SHOW	•	-	•	~	1430	412	129	37	~	, ma	•	25.1
	**************************************	•	~	211		3967	1927	784	370	176	\$	~	35.2
. 42	DIAMOND SOLO PRACTICE SHOW PRACTICE SHOW	•	2	141	418	2775	1218	794	170	;	•	•	31.8
~ I	NONO	•	•	•	:	1650	159	191	92	80	•	•	•••
	DIAMOND DIAMOND PRACTICE SHO	•	^	12	38	2506	104	105	*	~	•	•	30.0
		-3.00	-2.00	-1.00	٤.	•••							DATA
	O LEVEL	 LESS THAN -3.00	-2.99 10 -2.00	-1.90 TO -1.00	99 10	2.00 10	3.00 10	4.00 10	5.00 10	6.00 10	7.00 10	8.00 AND	HOURS OF DATA
		: 5	•	•		49							Í

TABLE XII OSCILLOGHAPH DATA UTILIZATION RATES PCR 1808 HOURS

	•		1	MADC	762	76-3	0						
	SOLO SOLO SOLO	1238	3676	6589	20285	105978	53577	21597	10037	3985	710	80	.00
1974-1975	SK PRACTICE	959	2634	6801	15622	03320	40174	17028	4909	1392	201	•	77
1974	DIAMOND PRACTICE SHOW PRACT	•	105	380	3734	76181	26730	7975	2743	823	160	•	**
	PRACTICE SH	-	22	***	203	70411	19291	4186	1601	191	18	•	55.9
	40HS 2HOHS	2193	2655	10592	21447	100789	\$2675	21075	9646	3202	205	;	•
1975	PRACTICE SHOW PRACTICE SHOW	1142	6644	\$140	17459	0396 1	29514	18881	1659	1352	163	•	45.9
-,	DI AMOND SHOW	•	•	72	3993	70252	22156	6511	1942	683	22	•	27.B
	PRACTICE	•	0	120	3267	56972	16414	\$139	1.47	279	•	c	25.1
	SOLO .	•	1193	2994	16776	112699	. 17/15	22273	10739	2000	1270	57	35.2
1974	PRACTICE SMOW PRACTICE SMOW	•	597	4623	13145	49264	38302	14520	5346	1667	252	•	31.0
~ •	DIAMOND CE SHOW	•	255	616	3367	26598	33214	10001	3678	1020	306	•	••
	PRACTICE	•	•	340	1039	91364	16136	3409	170	\$	•	•	30.
		.5.0	-2.00	-1.00	• 25	5.99	3.99	•	5.99	6.0	7.0	<u>a</u>	4
	e LEVEL	LESS THAN -3.00	-2.09 TO -2.00	-1.99 TO -1.60	02 66.	2.00 10	3.00 10	4.00 TO	5.00 70	6.00 10	7.00 70	8.00 AND UP	HOURS OF DATA

TABLE XIII OSCILLOGRAPH DATA CROSS COUNTRY CUMULATIVE EXCEEDANCES AND RATES PER 1000 HOURS

	CUMULA	21	E EXCEEDANCES	CUMULA	CUMULATIVE EXCEEDANCE RATES PER 1000 HOURS	EDANCE HOURS
	1974	1975	1974/75	1974	1975	1974/75
	•	:	• • • •	•	:	
LESS THAN -3.00	•	~	~	•	534	96
-2.99 TO -2.00	•	30	£	63	1003	438
-1.99 TO -1.00	32	55	87	670	1639	1121
99 70 .25	96	115	211	2012	3846	2719
2.00 TO 2.99	558	549	1107	11698	. 18361	14265
3.00 10 3.99	161	566	457	4004	8896	5889
4.00 TO 4.99	.	107	191	1761	3578	1976
5.00 70 5.99	7	35	76	628	1170	919
6.00 TO 6.99	11	~	54	356	234	309
7.00 10 7.99	~	-	~	20	33	52
8.00 AND UP	•	•	•	•	•	o
HOURS OF DATA	1.14	29.9	77.6	47.7	29.9	77.6

TABLE XIV

CSCILLOGRAPH VS. COUNTING ACCELEROMETER EXCEEDANCES

<u>YEAR</u> 1974	SERIAL NUMBER 154986	AIRCRAFT POSITION #4 DIAM	UTILIZATION PRACTICE SHOW	534 73 107	COUN ACCELE 6 6 22 28	TING ROMET 7 0 6	ER 8 0 0 0	5 21 71 92	0SC1 6 1 18 19	LLOGR. 7 0 4	<u>8</u> 0 0	OSC. HOURS 30.8 19.6 50.4
1974	155029	# 5 SOLO	PRACTICE SHOW	5 41 122 163	6 9 45 54	7 1 5 6	8 0 0	44 130 174	6 9 45 54	7 0 4	8 0 0	10.1 14.3 24.4
1974	154179	#6 SOLO	PRACTICE SHOW	5 72 106 178	6/6 41 47.	7 1 	8 0 1	58 58 98 156	6/4 39 43	7 0 4	8 0 1	17.6 <u>a.9</u> 27.5
1974	154177	∲6 SOLO	PRACTICE SHOW	5 48 147 195	29 90 119	7 5 26 31	8 0 0	5 45 135 180	28 84 112	7 5 26 31	8 0 0	4.1 11.0 15.1
1975	154975	#4 DIAM	PRACTICE SHOW	5 45 54 99	6 8 18 26	7 0 4	8 0 -1 1	5 36 51 87	6 7 16 23	7 0 2 2	80	25.1 27.8 52.9

TABLE XIV
OSCILLOGRAPH VS. COUNTING ACCELEROMETER EXCEEDANCES

<u>YEAR</u> 1975	SERIAL NUMBER 155029	AIRCRAFT POSITION #5 SOLO	UTILIZATION PRACTICE	<u>5</u> 76	CO ACCEL 6 17	UNTING EROMET 7 3		6 <u>5</u>	0SCILI 6 9	LOGRA1 7 1	<u>≥H</u> 8 0	osc. Hores
		77 0000	SHOW	123 199	51 68		1	116 179	46 55	7 8	<u>1</u>	$\frac{12.5}{26.3}$
1975	154176	∲6 SOLO	PRACTICE	24 4	4 <u>6</u>	7 5	8 0	214 214	4 <u>6</u>	7 3	<u>8</u>	29.1
			SHOW	390 634	123 171	<u>5</u>	_ <u>0</u>	305 519	94 138	<u>5</u> 8	<u>_0</u>	$\frac{33.1}{62.2}$

TABLE XV

DIAMOND AND SOLO SURVEY: PERCENT EXCEEDANCES IN ALTITUDE AND AIRSPEED RANGES

DIAMOND 1974-1975

PERCENT OF TOTAL EXCEEDANCES	ALTITUDE RANGE IN METRES (FT.)	PERO LESS THAN 300 KNOTS	CENT EXCEEDANCES LESS THAN 400 KNOTS	IN ALTITUDE LESS THAN 500 KNOTS	RANGE LESS THAN 600 KNOTS
50.6	LESS THAN 60.93(2000)	11.3	67.7	98.9	99.9
92.6	LESS THAN 1523.7(5000)	14.8	75.0	99.3	99.9
99.4	LESS THAN 3047.7(10000)	17.0	75.9	99.3	99.9
99.8	LESS THAN 4571.7(15000)	17.0	75.9	99.3	99.9
100.0	LESS THAN 9143.7(30000)	17.0	75.9	99.3	99,9

SOLO 1974-1975

PERCENT OF TOTAL EXCEEDANCES	ALTITUDE RANGE IN METRES (FT.)	· Peri Less than 300 knots	CENT EXCEEDANCES LESS THAN 400 KNOTS	IN ALTITUDE LESS THAN 500 KNOTS	RANGE LESS THAN 600 KNOTS
58.9	LESS THAN 609.3(2000)	5.6	57.9	98.7	99.9
87.2	LESS THAN 1523.7(5000)	12.5	67.5	99.0	99. 9
99.1	LESS THAN 3047.7(10000)	18.7	71.1	99.1	99.9
99.7	LESS THAN 4571.7(15000)	18.9	71.3	99.1	99.9
100.0	LESS THAN 9143.7(30000)	18.9	71.3	99.1	99.9

ACKNOWLEDGEMENT

The authors wish to acknowledge R.M. Catanese, W. Williams, A. Kaniss, and the Data Development Section of the Structures Division for their assistance in the preparation of this report.

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- (a) McDonnell Douglas Corporation Report Number, MDC J6698/01, "A-4FBA Blue Angel Wing Fatigue Analysis," March 1975
- (b) Century Model 409 Instruction, Century Electronics and Instruments, Fulsa, Oklahoma
- (c) Barber, Clyde, "Naval Flight Loads Research Program," Digest of United States Naval Aviation Electronics/TPD Booklet 7-65, September 1965
- (d) Naval Air Engineering Center Report No. NAEC-ASL-1106, Survey of Flight Load Parameters of Service Aircraft, Seventh Summary, 13 June 1967

APPENDIX A

INSTRUMENTATION DESCRIPTION

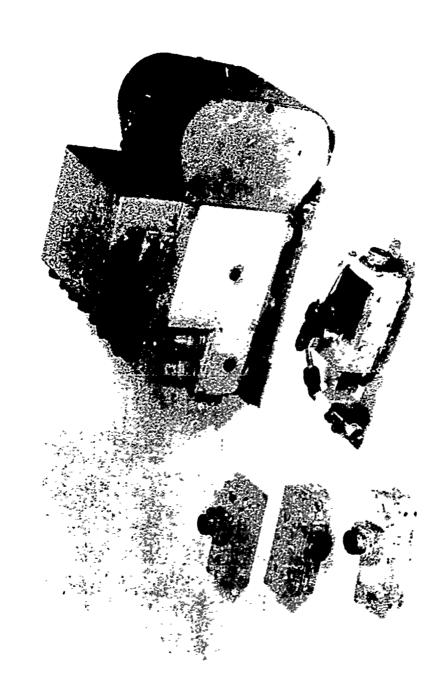
Instrumentation Description

The Century model 409 oscillograph system (Figure A-1), consisting of the oscillograph, bridge balance unit, pressure and acceleration transducers, and associated components, was designed for applications where space and weight requirements are critical, as well as for adverse shock, vibration, and temperature conditions. Weighing approximately 9.072 kilograms (20 pounds), this system is capable of continuously monitoring eight flight parameters, each simultaneously recorded on a photosensitive film or paper. (Figure A-2 shows the recording galvanometers used in this unit). In principle, the oscillograph records deviations of current flow in a galvanometer coil from those current values associated with level flight. Each flight parameter is associated with its own transducer and galvanometer. As the transducer changes the current flow in the galvanometer coil, motor actian deflects the mirror in a direction and amount determined by the transducer output. As the mirror is deflected, the trace of the light beam is recorded on the moving film or paper via the optical system shown in Figure A-3. Using a series of collimating mirrors, the light beam is narrowed until the projection of light from each galvanometer on the film (or paper) is a very small spot. When several flight parameters (galvanometers) are being recorded, necessary trace identification is made by momentarily interrupting each light path in sequence. Since the sequence of interruption is independent of the trace position on the film (or paper), it is always possible to determine which galvonometer is responsible for a particular trace. The rate of interruption varies with film/(paper) speed, so regardless of that speci, a trace is interrupted the same number of times in a given length of record. Magazines of the light sensitive film, called records, are 45.72 metres (150 feet) long and are capable of recording approximately five hours of flight time (the oscillograph is activated only in the wheels up configuration) at a rate of 0.1524 metres (6 inches) per minute. Oscillograph units for the A-4F Blue Angel flight usage data study provided a continuous time history of aircraft airspeed, altitude, and normal acceleration. The recording portion of the oscillograph in all instrumented aircraft was located at fuselage station 20 in the nose compartment. Airspeed and altitude transducers were located approximately at fuselage station 30.0. The normal acceleration transducer, mounted on rigid structure close to the aircraft center of gravity to eliminate angular accelerations, was located at fuselage station 236.0 port side frame.

Systron-Donner counting accelerometer units (Figure A-4) weighing approximately 2.268 kilograms (5 pounds), provide an automatic, in-flight (wheels-up) permanent record of normal acceleration at four pre-set load or "g" levels. The transducer is a solid state, force-balance device whose output voltage is indicative of the induced magnetic field voltage required to return a mass to its neutral (level flight) position. Output voltage is registered by the indicator via window counters, with each window corresponding to a different load/g level. By taking periodic readings and comparing with previous readings, the cumulative number of exceedances at each level can be determined for the time interval between readings.

Counting accelerometer units instalied in Blue Angel aircraft for this study measured normal accelerations at the 5.0, 6.0, 7.0, and 8.0g levels. These units are referred to as "dash seven" type. MIL SPEC A-22145BAS (February 28, 1970) requires the counting accelerometer to operate within +0.1g at the 5.0 and 6.0g levels, and within +0.15g at the 7.0 and 8.0g levels. Transducer serial numbers and calibration levels for Blue Angel aircraft are indicated in Table A-I. The transducers were located at fuselage station 236.0 port side frame, beside the oscillograph transducer. Indicators were located in the portside landing gear wheel weil.

Figure A-1. Oscillograph System



APPROX. 30.5cm. X 15.7cm. X 15.7 cm.

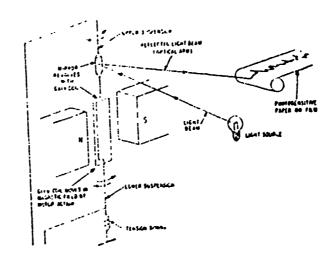


FIGURE A-2: GALVANOMETER UNIT

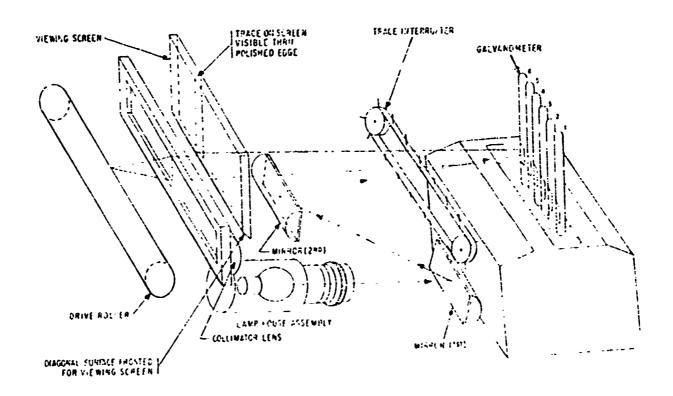


FIGURE A-3: OSCILLOGRAPH OPTICAL SYSTEM

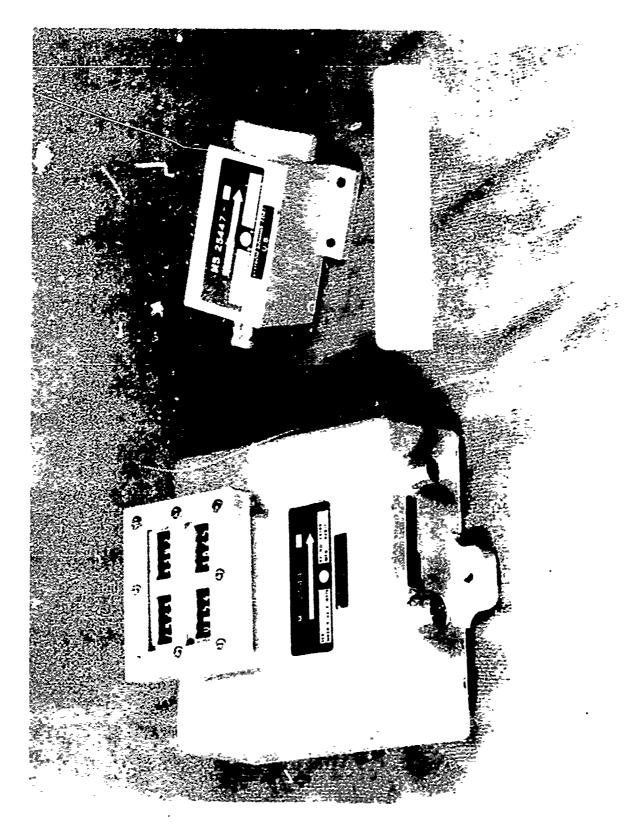


Figure A.4. Counting Accelerometer System

TABLE A-I
TRANSDUCER SERIAL NUMBER AND CALIBRATION LEVELS

SERIAL			<u>C</u>	NLIBRAT	ION LEVI	eis
NUMBER	TRANSDUCER	DATE INSTALLED	5	<u>6</u>	7	<u>8</u>
154174	CXY 5171	3/6/74	5.03	6.01	7.04	8.07
	CXY 5140	4/2/75	4.99	6.01	7.08	8.02
	CXY 5181	8/26/75	5.09	6.08	7.08	8.07
154177	CXY 5187	3/6/74	5.09	6.08	7.12	8.11
154179	CXY 5170	3/6/74	5.03	6.01	7.04	8.07
	CXY 0941	3/23/74	5.09	6.08	7.15	8.15
154975	CXY 5183	3/6/74	5.06	6.03	7.08	8.07
154983	CXY 5188	3/6/74	5.06	6.04	6.99	8.00
	CXY 5157	5/16/74	5.06	6.08	7.08	8.10
154984	CXY 5186	3/6/74	5.09	6.08	7.08	8.11
154986	CXY 5178	3/6/74	5.09	6.08	7.12	8.15
	CXY 5177	1/15/75	5.03	6.01	7.02	8.01
155029	CXY 5124	3/6/74	5.03	6.05	7.04	8.07

APPENDIX B

BLUE ANGEL AIRCRAFT COUNTING ACCELEROMETER

DATA BY UTILIZATION FOR 1974 AND 1975

1974

UTILLIZATION - BLUE ANGLI COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 154176

UTILIZATICH	HOURS	ACCEPTABLE COUNTING ACCELERONETER	ın l	O E S C E E D	EDANCES	ωl
PRACTICE DIAMOND	15.8	15.8	54	ო	0	0
PRACTICE SOLO	14.1	14.1	37	ო	0	0
SHOW DIANOUD	11.6	11.6	54	Ŋ	0	0
опоя моня	0.7	0.7	15	Ŋ	0	0
CROSS COLUTAY	83.1	83.1	23	Ŋ	H	-
CHECK OUT	11.2	11.2	10	8	0	Ö
OT:ER	0.0	0.0	t	ŧ	•	1
UNSTORE	8.99	19.5	5	0	0	0
TOTALS	203.3	156.0	142	23	1	-

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UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

1975

154176
NUMBER:
SERIAL

UTITISATICH	HOURS	ACCEPTABLE COUNTING ACCELEROMETER	νl	ы х с в х с в х з	EXCEEDANCES	ωl
PRACTICE DIAMOND	1.3	1,3	9	0	0	0
PRACTICE SOLC	188.0	142.1	1394	369	28	0
SHOW DIANCHD	0.0	0.0	ı	•	ŧ	ı
SHC% SOLO	52.0	46.0	541	202	13	0
CAUSS COLUTRY	141.3	125.9	122	27	8	0
כאבכא פתו	6.3	5.0	8	1	0	0
OTHER	5.4	5.4	0	0	0	0
CNKNOSS	0.0	0.0	•	t	•	1
R tals	394.3	325.7	2065	299	43	0

1974

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 154177

UTILIZATIC:	HOURS FI.OKN	ACCEPTABLE COUNTING ACCELERONETER	ınl 	EXCEE	EXCEEDANCES	ωl
PRACTICE DIAMOND	21.5	21.2	16	7	0	0
PRACTICE SOLO	24.9	24.9	161	96	27	4
SHOW DIAMOND	7.6	6.7	7	0	0	0
SHOW SOLO	33.8	33.8	426	242	70	m
GROSS COLLITRY	122.6	122.6	89	24	4	0
כאבכה כנד	6.1	6.1	∞	ĸ	0	0
OTHER	0.0	0.0	•	ì	1	1
ישיטנינית	121.5	9.7	7	7	8	0
TOTALS	340.1	228.0	723	372	103	7

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UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

1975

SERIAL NUMBER: 154177

UTTTTSVIIO:	HOURS	ACCEPTABLE COUNTING ACCELERONETER	inl	ы К С С С С С	ZXCEEDAXCES	œΙ
PRACTICE DIAMOND	106.5	97.6	92	15	4	1
PRACTICE SOLO	26.9	23.2	163	23	0	, · O
SHOW DEAMOND	47.5	43.1	78	30	12	ო
SHOW SOLO	0.0	0.0	1	8	ı	ı
CROSS COUNTRY	131.0	122.7	۲	~ !	0	0
CHECK OUT	5.1	5.1	4	0	0	0
oth e k	8.2	8.2	74	0	0	0
CONTROLAT	2.6	2.6	0	0	0	0
TOTALS	327.8	302.5	352	69	16	7

1974

TTILIZATION - BIME ANGEL COUNTING ACCELERONETER DATA

SERIAL NUMBER: 154179

	HOURS	ACCEPTABLE COUNTING ACCELERONETER	ıΛΙ	EXCEEDANC 6	ν ν ο ι · :	∞l
PRACTICE BIANOUD	2.0	2.0	.	0	0	0
PRACTICE SOLO	58.3	41.4	194	25	ന	0
SHOW DIAMOUD	1.7	1.7	4	0	0	0
970S %3HS	25.6	25.6	242	97	11	.7
CROSS COCHRY	70.1	65.8	32	9	7	0
THECK OUT	***	4.4	10	1	0	0
OTHER.	0.0	0.0	•	,	ı	ı
13.50.0MS	138.2	19.3	7	7	0	0
Sivil	300.3	160.2	067	131	16	2

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		4 + 4 + 4 + 4 + 4 + 4 + 4 + 4 + 4 + 4 +				
CILIZATION:	HOURS	ACCELERONETER	νl	EXCEEDANC 6	DANCES	ωi
PRACTICE DIAMOND	196.0	196.0	366	84	12	0
PRACTICE SOLO	5.2	5.2	18	10	7	-
SHOW DIAMOND	51.6	51.6	102	34	œ	0
SHOW SOLO	0.0	0.0	•	•	ı	ŧ
CAUSS COLITRY	149.1	149.1	13	ฑ์	1	0
CHECK OUT	3.7	3.7	x 0	0	0	0
uhito	8.8	8.8	0	0	0	0
e e e e e e e e e e e e e e e e e e e	1.6	1.6	0	0	0	0
TOTALS	413.0	413.0	507	131	25	-

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

1975

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UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

1974

	HOURS	ACCEPTABLE COUNTING ACCELERONETER	νl	EXCEEDANCES	N C C C C C C C C C C C C C C C C C C C	ωl
PRACTICE DIAMOND	86.2	77.5	81	7	0	0
PRACTICE SOLO	0.0	0.0	•	•	•	•
SHCW DIAMOND	7.67	47.2	76	ω	انتو	0
SHC:: SOLO	0.0	0.0	•	•	ı	•
CROSS COUNTRY	112.3	108.9	Ŋ	0	0	0
CHECK OUT	9.4	9.4	0	o	o	0
OTHER	1.2	1.2	0	0	o	0
UNICONIA	26.8	6.4	0	0	0	0
TOTALS	280.5	244.3	180	15	1	0

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UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

1975

CTITIZATIC:	HOURS	ACCEPTABLE COUNTING ACCELERONETER	ı∩l	ENCEEDANC 6 2	DANCES	ωi
PRACTICE DIAMOND	192.3	192.3	329	06	12	0
PRACTICE SOLO	6.7	6.7	09	54	9	0
SHOW DIANOND	55.2	55.2	104	33	4	H
0708 %088	0.0	0.0	1	•	•	•
CROSS COUNTRY	151.3	151.3	01	1	0	0
מובכא סנד	1.9	1,9	•	က	0	0
07:23	2.6	2.6	Ó	•	0	0
:	0.0	0.0	•	•	ı	•
TC::41.S	410.0	410.0	209	151	22	1

1974

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 154983

UTILIZATIC::	HOURS	ACCEPTABLE COUNTING ACCELEROMETER	νl	ENCEEDANC 6	요 이 (시 오	ωI
PRACTICE DIAMOND	84.2	84.2	20	30	e	0
PRACTICE SOLO	3.4	3.4	0	0	0	o
SHOW DIAMOND	38.5	38.5	39	27	9	0
SHOT SOLO	0.0	0.0	•		•	ı
CROSS COUNTRY	99.1	99.1	m	8	0	0
CHECK OUT	7.8	7.8	0	0	0	0
OTHER	5.0	5.0	0	0	0	0
UNGTOWN	133.3	7.6	0	0	0	0
TOTALS	371.3	245.6	92	59	6	0

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ωl

1975

RONETER DATA	EXCEEDANCES 6 7	62 10 0	43 12 1	0 01 07	•	1 0 0	1 0 0	2 0 0	•	149 32 1
UTILIZATION - BLUE ANGEL COUNTING ACCELERONETER DATA SERIAL NUMBER: 154983	ACCEPTABLE COUNTING ACCELERONETER	91.9	0.9	43.6	0.0	125.8	5.4	12.5	0.0	285.2
UTILIZATION	HOURS	91.9	0.9	43.6	0.0	125.8	5.4	12.5	· · · · · · · · · · · · · · · · · · ·	285.2
	CTILIZATION	PRACTICE DIAMOND	PRACTICE SOLO	SHOW DIANOND	otos 2018	CROSS COUNTRY	CHECK OUT	OTHER	UNCOME	Totals

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DATA	
: - BLUE ANGEL COUNTING ACCELEROMETER DATA	
COUNTING	
AXGEL	
BL:: E	
TILIZATIO:: -	

.

:::::ZATIO::	HOURS	ACCEPTABLE COUNTING ACCELERONETER	νl	EXCEEDANCE	DANCES	ωl
PRACTICE DIWOND	93.9	81.9	9	'n	0	•
PRACTICE SOLO	0.0	0.0	•	•	•	•
SHOW DIAMOND	6.04	40.9	22	0	0	0
070S :: 285	0.0	0.0	•	•	•	•
CADSS COURTAY	113.3	113.3	•	0	0	0
CHICK OUT	6. 5	6.5	'n	Ö	ó	0
0	0.0	0.0	1	•	•	•
:::0::::	110.9	3.2	0	0	0	0
\$71172	365.5	245.8	93	\$	0	0

1975. UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

154984

SERIAL NUMBER:

ωl EXCEEDANCES 6 7 2 17 46 104 11 35 134 747 24 Ö 88 161 Ò 2 ACCEPTABLE COUNTING ACCELERONETER 0.0 325.2 144.8 102.4 2.2 1.6 44.7 11.7 17.8 102.4 144.8 2.2 0.0 1.6 325.2 HOURS 44.7 11.7 17.8 PPACTICE DIAMOND PRACTICE SOLO CANS COUNTRY SHOW DIAMOND U. IZATION SHOW SOLO CHECK OUT 230:23:3 TCIMIS OTHER

NADC 76276-30

1974

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 154986

101111111	HOURS FLOWN	ACCEPTABLE COUNTING ACCELERONETER	ınl	ENCEEDAN 6	м о м о	ώl
CHRETE BITTORE	83.0	83.0	105	23	m	0
PRACTICE SOLD	0.0	0.0	•	•	•	•
enowic moses	42.6	42.6	991	55	23	1
01C5 :::3HS	0.0	0.0	•	•	•	•
Catas country	118.8	118.8	v	R	-	0
The Meridian	2.5	2.5	0	0	0	•
	10.1	10.1	TI.	ju .	0	0
	14.6	1.7	0	0	0	0
70.318	. 271.6	258.7	288	81	27	7

NADC 76276-30

with the same of t

1975

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

UTILIZATIC:	HOURS FLOWN	ACCEPTABLE COUNTING ACCELEROWETER	ni	E X C E E	EXCEEDANCES	∞l
PRACTICE DIANOND	150.7	150.7	221	79	16	ń
PRACTICE SOLO	20.6	20.6	132	67	ñ	0
SHOW DIANOND	12.7	12.7	Õ	₩	Ħ	اش
otos aois	1.5	1.5	16	vo	0	Ó
CROSS COLTITRY	59.5	\$.65	'n	၈	8	8
CHECK OUT	e,	3.3	17	œ	ò	0
OTHER	ò.2	5.0	•	0	o	Õ
באַניסייב:	3.7	3.7	0	0	0	0
TOTALS	257.0	257.0	421	132	22	قة

1974

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 155029

UTITZATIO:	HOURS	ACCEPTABLE COUNTING ACCELEROMETER	اد.	EXCEEDANC 6	ANCES	ωI
PRACTICE DIAMOND	1.2	1.2	4	1	0	0
PRACTICE SOLO	83.6	83.6	456	75	ω	Ō
SHOW DIAMOND	0.0	0.0	•	•	ı	ı
OTOS MOHS	41.7	41.7	383	136	10	0
CROSS COUNTRY	1uó.2	106.2	75	18	0	0
CHECK OUT	6.3	6.3	٠	ю	o	0
OTHER	3.1	3.1	0	0	0	Õ
UNECORN	44.1	1.4	∞	4		0
TOTALS	286.2	243.5	932	23.7	19	0

1975

UTILIZATION - BLUE ANGEL COUNTING ACCELEROMETER DATA

SERIAL NUMBER: 155029

UTTTEATION	HOURS	ACCEPTABLE COUNTING ACCELEROMETER	ıñl 	E X C E E	EDANCES	اف
PRACTICE DIAMOND	0.0	0.0	t	1	ı	1
PRICTICE SOLO	136.0	136.0	1023	218	. 61	Ā
SHOW DIAMOND	0.0	0.0	•	ı	i	ı
SHOW SOLO	36.5	36.5	587	180	36	9
CROSS COUNTRY	102.6	102.6	75	22	2	0
CHECK OUT	5.1	5.1	7	9	1	0
OTHER	3.9	3.9	0	0	Ö	0
CNEWOWN	1.2	1.2	, ,	0	0	0
TCTALS	285.3	285.3	14,92	426	88	7

APPENDIX C

SURVEY OF AIRSPEED, ALTITUDE, AND EXCÉEDANCE FOR 1974-1975 DIAMOND AND SOLO A4F BLUE ANGELS 1974-1975 DIAMOND

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TABULATION

ALTITUDE -2000 TO -1 FEET (-609.60 TO -.30 METRES) AVERAGE GROSS WEIGHT 15110 POUNDS (6853.66 KILOGRAMS)

				ä	EQUIVALENT AIMSPEED. KNOTS	ENT ATH	SPEED	KNOTS	, ۸			•	
LOAD FACTOR	10R	100	150 100 199	200 10	250 10 299	349	350 10	4 40	450 499	500 100 100 100	550 10 599	649 649	TOTAL
LESS THAN -3.00	ā.50	•	•	•	•	ø	•	•	•	•	•	•	•
-2.99 TO -2.00	2.00	•	•	•	•	•	•	•	•	0	•	•	•
-1.99 TO -1.00	00.1	•	•	•	•	•	•	٥	•		•	•	•
99 10	•25	•	•	•	•	М	~	•	•	•	٥	•	. ທ
2.00 10	5.99	•	•	•		12	68	7	21	4	0	0	185
3.00 TO	3.99	۰	•	•	•	•	~	10	S	•	.•	•	22
4.00 TO	66.5	•	•	.0	•	•	~	•	•	•	•	•	~
5.00 TO	5.99	•	•	•	•	•	•	•	•	**	•	٥	-
6.00 TO	66.9	٥	•	•	•	•	•	•	•	•	•	0	0
7.00 10	7.99	•	•	•	•	•	•	•	•	•	•	•	•
A.00 AND	a n	•	•	•	•	•	•	•	•	•	•	•	•
TOTAL EXCEEDANCES	ANCES	0	٥	0	-	24	4	8	56	v	0	0	216

A4F BLUE ANGELS 1974-1975 DIAMOND

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT ATRSPEED TABULATION

ALTITUDE 0 TO 1999 FEET (.00 TO 609.30 METRES)

AVERAGE GROSS WEIGHT 14569 POUNDS (6651-67 KILOGRAMS)

EQUIVALENT ATMSPEED, KNOTS

				j			7775	CODIVALENI MINSPEEDO ANDIS	0				
		100	150	200	250	300	150	604	450	500	550	600	600
LOAD FACTOR	or	149	199	543	568	349	366	440	664	540	299	649	TOTAL
LESS THAN -3.00	0	•	•	•	•	•	•	•	٥	0	•	•	
-2.99 TO -2.00	00	•	ĸ	۰	-	•	٥	•	•	•	•	•	
-1.99 TO -1.00	00	•	•	٥	۰.	-	-	•	0	•	¢	•	••
99 10	•25	•	•	0	Ξ	*	36	. v	~	~	0	•	č
2.00 10 2.99	6	•	8	26	394	1064	1149	732	217	18	n	m	3674
3.00 TO 3.99	6	•	0	=	55	102	401	334	112	^	6	•	1029
4.00 TO 4.99	66	•	•	•	23	31	58	72	4.	4	•	9	234
5.00 TO 5.99	6	•	٥	•	~	~	=	15	•	r	~	0	3
6.00 TO 6.99	\$	•	•	•	0	-	•	Ŋ	4	-	•	0	-
7.00 10 7.	7.99	•	•	0	•		•	~	~		•	•	•
A.00 AND UP		0	•	•	•	•	•	0	•	c	0	•	
TOTAL EXCEEDANCES	CES	٥	N	106	491	1231	1662	1165	389	36	œ	m	509

AAF BLUE ANGELS 1974-1975 DIAMOND

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TARINLATION

ALTITUDE 20"0 TO 4499 FEET (609.50 TO 1523.70 METRES) AVERAGE GROSS WEIGHT 14634 POUNDS (6637-94 KILOGRAMS)

COUIVALENT AIMSPEED, MNOTS

	•	-	-	2	000	* ·	7661	*	133	Ξ	•	DANCES	ក្ល	TOTAL EXCEEDANCES
•	•	•	C	•	0	•	•	0	•	•	•	d C	Z	A.00 AND UP
m	•	•	_		-	•	•	•	•	•	•	7.99	10	7.00 TO
20	•	•	c	~	~	10		•	•	•	•	4.49	13	6.30 10
112	۰	C	c	•	35	36	58	6	-	•	•	8.99	10	5.00 TO
270	•	~	-	å.	2	5	88	15	-	•	•	66.4	Ç.	4.00 10
983	•	C	r	51	221	373	180	4	^	•	•	3.99	13	3.00 19
2947	•	C	•	4	219	758	1243	404	117	•	0	5.49	2	2.00 10
118	•	•	•	•	•	3	35	36	N)	S	•	•25	10	01 40
æ	•		c	•	•	•	•	•	•	•	3	-1.00	10	-1.99 TO -1.00
*	•	◦.	•	•	•	•	-	•	•	•	•	-2.00	10	-2.99 TO -2.00
•	•	E	•	•	•	•	•	•	•		•	-3.00	Z	LESS THAN -3.00
TOTAL	\$ 100 100 100 100 100	550 10 599	500 10 549	450 10 499	4 4 4 4 6 4 7	350 10 199	349 349	250 10 299	200 249	150 10 199	1,00	LOAD FACTOR	ī	L0A0

ANF BLUE ANGELS 1974-1975 DIAMOND

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEEN TABULATION

ALTITUDE 5000 TO 9999 FEET (1524-00 TO 3047-70 METRES) AVERAGE GROSS WEIGHT 14456 POUNDS (6556.97 KILOGPAMS)

				2	UTVALE	INT AT	EQUIVALENT AIMSPEED, KNOTS	KWOT				•	
LOAD FACTOR	000		150 109	249 70 249	250 10 299	340	350 399	4 4 00 4	450 499	500 100 540	550 10 599	60 44 60 64 64 64	600 TO 649 TOTAL
LESS THAN -3.00	0	_	•	ė	•	0	. •	0	9	•	•	٥	٥
-2.99 TO -2.00	9	_	۰	•	•	m	*	٥	•	c	•	•	~
-1.99 TO -1.00		_	•	~	•	=	~	•	0		•	0	20
99 70 .25		 	=	2	58	88	16	m	М	•	•	•	140
2.00 10 2.99	6	_	S	\$	150	\$	9	=	•	•	-	•	360
3.00 TO 3.99	9	_	•	~	=	27	4 3	61	62	~	,•	•	129
4.00 TO 4.99	•	_	٥	.0	•	15	•	v	20	~	0	0	45
5.00 TO 5.99	9	_	0	•	0	~	'n	w	~	-	٥	•	1
6.00 TO 5.99	9	_	•	•	•	•	**	v	~	c	0	c	æ
7.00 TO 7.99	9	_	•	•	c	Ö	•	•	•	c	0	•	•
4.00 AND 11P		_	•	•	c	٥	•	•	0	c	0	•	•
TOTAL EXCEEDANCES	ie S	_	36	2	231	184	119	4	e 4	4	-	0	719

1974-1975 DIAMOND A4F BLUE ANGELS OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TAHULATION

ALTITUDE 10000 TO 14499 FEET (3048.00 TO 4571.70 METRES)

AVERAGE GROSS WEIGHT 15249 POUNDS (6938.96 KILOGRAMS)

EQUIVALENT ATHSPEED, KNOTS

				,									
LOAD FACTOR	30 ¥0 ¥		150	200 240 240	250 70 299	369	350 10 399	\$ 1 \$ 0 \$ 0 \$ 0 \$ 0	450 499	500 549	550 10 599	\$ 40 0 10 0 10 0 10	TOTAL
LESS THAN -3.00	.3.00	•	•	•	•	•	•	۰	•	c	•	•	•
-2.99 TO -2.00	.2.00	•	•	•	•	•	•	•	٥	•	•	•	•
-1.99 TO -1.00	1.00	•	•	•	۰.	•	ó	•	•	•	•	•	•
99 10	.25	0	•	•	•	-	13	: -	~	c	•	. 0	~
2.00 10	5.09	٥	•	-	*	~	v	~	~	c	•	0	20
3.00 To	3.99		•	•	~	C	•	c	-	c	•	0	æ
4.00 10	60.4	•	•	۰	-	-	-	•	۰	•	Ç	•	m
5.00 TO	5.99	•	۰	•	•	•	•	•	•	•	•	•	•
6.00 TO	6.9	•	•	•	•	•	•	•	•	•	•	•	•
7.00 10	7.99	•	•	•	•	0	•	9	•	c	•	٥	ပ
4.00 AND UP	<u>a</u>	۰	•	•	•	•	•	•	•	c	•	•	•
TOTAL EXCEEDANCES	ANCES	•	•		~	15	16	~	4	c	0	0	4 75

A4F BLUE ANGELS 1974-1975 DIAMOND

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT A: RSPEED TABULATION

ALTITUDE 15000 TO 29999 FEET (4572.00 TO 9143.70 METRES)

AVERAGE GROSS WEIGHT 15667 POUNDS (7106-28-KILOGRAMS)

EQUIVALENT ATHSPEED, KNOTS

				20 I VAL.	- Z	EQUIVALENT ATHSPEED, KNOTS	X X 0 1					
LOAD FACTOR	201	150 199	200 240 240	250 250 250	300 240 240	350 10 399	9-4	\$ 100 \$ 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	800 840 840	550 10 549	6014	101
LESS THAN -3.00	•	•	•	•	۰	0	•	•	•	•	•	
-2.94 TO -2.00	٠	•	•	•	•	•	•	•	c	٥.	•	
-1.99 70 -1.00	•	•	•	•	⊖ '	•	•	•	¢	•	0	
99 TO .25	•	•	•	•	•	m	•	•	•	•	•	
2.00 10 2.99	۰	~	٠	•	•	~	-	•	•	•	9	
3.00 10 3.49	•	•	•	~	•	•	•	•	c	0	•	
4.00 TO 4.99	•	•	•	•	•	•	•	•	•	•	•	
5.60 TO 5.99	•	•	•	•	•	•	•	•	•	•	0	
6.00 TO 6.99	•	•	•	•	•	•	•	•	•	•	0	
7.00 TO 7.99	•	•	•	•	•	•	9	9	c	6	•	
A.OO AND 11P	•	•	•	•	•	•	•	•	•	0	0	
TOTAL EXCEEDANCES	•	-	•	~	n	s	-	•	•	•	•	_

. A4F BLUE ANGELS 1974-1975 SOLO

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TARULATION

ALTITUDE -2000 TO -1 FEET (-609,50 TO -.30 WETRES)

AVERAGE GROSS WEIGHT ISIZI POUNDS (6858-71 KILNGRAMS)

				ä	WIVALE	EQUIVALENT AIRSPEED. KNOTS	SPEED.	KNOTS				•	
LOAD FACTOR	0108	00 1 00 1	150	200 240 240	250 10 299	300	350 10	00+4 00+4	400 400 400	50 T S	550 549	\$ 4 \$ 0 \$ 0 \$ 0	101'AL
LESS THAN -3.00	-3.00	ō	۰	٥	0	•	ō	9	٥	c	0	•	•
-2.94 TO -2.00	-2.00	ō	•	•	۰	٥	ō	-	•	•	•	۰	~
-1.94 TO -1.00	-1.00	•	•	•	•	•	n	~	ò		•	•	•
99 TO	-25	٥	•	•	-	-	4	~	٥	•	•	•	13
2.00 10	5.00	•	٥	~	•	~	13	9	-	e	0	•	39
3.00 TO	3.44	•	•	•	~ :	_	11	=	~	c	. •	•	S
4.00 TO	60.4	•	•	.0	•	*	*	10	-	c	•	•	÷
5.00 10	5.99	0	0	•	•	•	•	•	٥	•	•	•	•
6.00 To	6.49	٥	•	٥	•	•		æ	~	~	•	•	æ
7.00 10	7.99	9	0	0	٥	•	•	•	•	•	•	٥	۰
4.00 AND UP	ā	•	•	•	0	•	•	9	•	c	•	0	•
TOTAL EXCEEDANCES	DANCES	0	c	-	Ξ	6	0	S	S		c	•	127

A4F BLUE ANGELS 1974-1975 SOLO

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TABULATION

ALTIT(10E 0 TO 1999 FEET 10 TO 109.30 HETRES)

AVERAGE GROSS WEIGHT 14776 POUNDS (6702-34 KILNGRAMS)

COUIVALENT AIMSPEED. KNOTS

				ı		i							
		100	150	200 100	250	300	350	40	450	500	550	200	\$00 10
LOAD FACTOR	5	149	100	546	290	349	399	744	56.7	540	299	649	TOTAL
. LESS THAN -3.00	00.	•	•	•	•	2	106	\$		•	•	9	13
-2.99 TO -2.00	.00	0	•	-	c	•	280	312	ř	^	ō	•	3
-1.99 TO -1.00	00.	~	•	•	₹.	23	345	. 405	50	0	•	•	è
99 10	.25	•	9	23	Ñ	į	614	420	ë	=	m	. •	102
2.00 10 2	2.99	•	~	6	553	1284	1877	1358	243	57	۴	•	2464
3.00 TO 3	3.99		•	9.	159	535	1510	1283	323	Š	S.	•	387
4.00 Tc 4	4.99	•	•	۰	=	166	# 0	427	.171	37	€	**	238
5.00 10 5	5.49	•	•	•	-	39	628	503	75	c	c	•	125
5.00 TO 6	6.49	•	•	•		W	214	244	•	C	~	٥	267
7.00 10 7	7.99	•	•	•	•		15	55	13	-	•	•	ž
A.00 AND U	ā	•	•	•	0	•	•	-	~		c	•	•
TOTAL EXCEEDANCES	NCES	N.	~	137	760	2153	6253	2602	926	183	52		1604

ANF BLUE ANGELS 1974-1975 SOLO

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TABULATION

ALTITUDE 2040 TO 4999 FEET (609.60 TO 1523.70 METRES)

AVERAGE GROSS WEIGHT ISIZE POUNDS (6861-23 KILOGRAMS)

EQUIVALENT ATHSPEED, KNOTS

			I									
	901	150	200	250	300	350	90	450	500	550	95	
LOAD FACTOR	149	\$	543	562	9	399	644	400	240	849	649	TOTAL
LESS THAN -3.00	•	•	•	-	•	12	~	•	•	•	•	54
-2.94 TO -2.00	•	•		4	m	2	•	•	ē	•.	•	ň
-1.99 70 -1.00	•	•	2	\$	23	35	9	***	c		•	124
94 10 .25	~	•	55	132	292	218	\$	•	c	•	•	17.
2.00 TO 2.99	•	75	301	1175	1392	781	274	•	•	-	•	3994
3.00 10 3.49	•		2	305	781	630	243	\$5	•	c	•	2042
4.60 TO 4.49	•	•	•	<u> </u>	177	942	112	:	•	c	-	571
5.00 10 5.99	•	•	•	-	=	2	;	•	~ :	c	•	136
4.00 10 6.99	۰	0	•	•	•		•	~	•	c	Ö	23
7.00 10 7.99	•	•	۰	•	•	'-	•	-	É	c	•	•
A.00 AND 11P	•	•	٠	•	•	•	-	•	c	c	•	~
TOTAL EXCEEDANCES	~	23	397	1691	2683	2024	A21	143	~		-	1768

. A4F BLUE ANGELS 1974-1975 SOLO

CASERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TABULATION

ALTITUDE 5000 TO 9999 FEET (1524.00 TO 3047.70 METRES) AVEHAGE GROSS WEIGHT 152A6 POUNDS: (6933-60 KILDGRAMS)

			ā	SUIVAL.	ENT AT	EQUIVALENT ATHSPEED, KNOTS	KNOTS					
	007	150	200	250	900	350	007	450	500	550	9	009
LOAD FACTOR	149	199	249	566	340	399	644	6	240	899	3	TOTA
LESS THAN -3.00	•	•	•	•	•	•	•	•	•	•	•	
-2.99 TO -2.00	•	•	~	*	•	•	9	•	•	•	•	
-1.99 TO -1.00		•	62	ë †	S	2	-	•		•	•	12
99 TO .25	**	2	174	544	135	39	٠	-	•	•	•	69
2.60 TO 2.99	~	2	383	932	417	9	*2	~	E	•	•	179
3.00 TO 3.99	•		22	223	204	2	36	n	c	.•	•	56.
4.00 TO 4.49	•	•	۵	1	37	25	*	~	c	•	•	ě
5.00 70 5.99	•	•	•	•	•	~		•	•	•	0	-
6.00 TO 6.99	•	•	•	•	•	~	•	•	•	•	•	
7.00 10 7.49	•	•	•	•	•	•	Ö	•	•	•	0	
4.00 AND 11P	٥	9	•	•	c	•	•	•	•	•	•	
TOTAL EXCEEDANCES	81 S:	\$	643	1359	798	273	2	~	•	•	0	326

A4F BLUE ANGELS 1974-1975 SOLO

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEEN TARULATION

ALTITUDE 10000 TO 14999 FEET (3048.00 TO 4571.70 METRES)

AVERAGE GROSS WEIGHT 15429 POUNDS (6998-68 KILOGHAHS)

EQUIVALENT ATHSPEED. KNUTS

90
149 199 249
•
0
•
2 1 11
0 1 11
0
•
•
•
0 0
0 0
2 2 22

A4F BLUE ANGELS 1974-1975 SOLO

OBSERVED NORMAL LOAD FACTOR VS. EQUIVALENT AIRSPEED TABULATION

ALTITUDE 1500C TO 29999 FEET (4572.00 TO 9143.70 METRES) AVERAGE GROSS WEIGHT 15127 POUNDS (6A61.62 KILOGRAMS)

EQUIVALENT AIMSPEED, KNOTS

TOTAL	0	v	•	27	20	S	0	•	٥	0	0	99
600 100 649	0	•	0	•	•	0	0	0	•	0	0	0
550 10 599	•	0	. •	•	•	•	٥	0	•	•	•	0
500 549	•	c	•	•	c	•	•	•	С	c	c	0
450 499	0	0	6	•	•	•	0	0	•	•	•	•
4 400 4 400 4 400	9	0	c	•	-	0	0	•	0	0	•	~
350 10	0	•	~	12	•	-	•	0	•	•	•	28
349	•	-	S	° cc	•	~	•	c	•	•	•	52
250 10 299	0	•		•		~	0	•	•	•	•	2
200 70	•	0	-	0	•	•	0	۰	•	•	•	~
150 109	•	•	•	•	٥	•	0	•	•	•	•	0
149	٥	•	•	-	•	•	٥	٥	•	•	٥	-
LOAD FACTOR	LESS THAN -3.00	-2.99 TO -2.00	-1.99 TO -1.00	.25	5.49	3.99	60.4	5.99	66.9	7.99	4.00 AND UP	TOTAL EXCEEDANCES
Ž.	Z	10	2	5	10	3.00 TO	2	5.00 TO	10	10	AND	CEE
0 4 0	<u></u>	36	0	99 10	2.00 TO	00	4.60 10	00	6.00 TO	7.00 10	00	Ą
د	LESS	-2.	7	i	٧.	ů.	;	ů.	¢	7.	*	TOTAL